

TECHNICAL DEPT.

21 1928

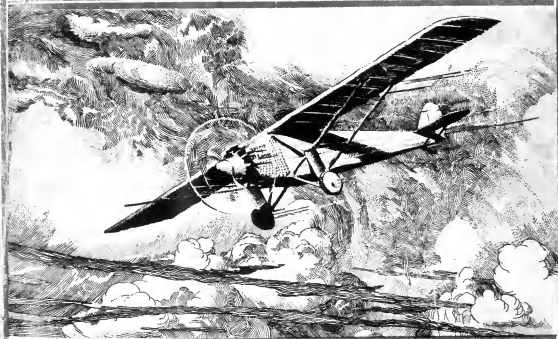
AVIATION

The Oldest American Aeronautical Magazine

MAY 21, 1928

Issued Weekly

PRICE 20 CENTS



Reproduction of an etching of the "Spirit of St. Louis" by John MacGilechrist

VOLUME.
XXIV

Special Features

The Curtiss "Robin"
The Hallett Aircraft Engine
Etching of Aluminum Alloy Propellers

NUMBER
21

TOLEDO
PUBLIC
LIBRARY

AVIATION PUBLISHING CORPORATION
250 WEST 57 STREET, NEW YORK

Publication Office, Highland, N. Y. Entered as Second-Class Matter, Nov. 25, 1920, at the Post Office, at Highland, N. Y. under Act of March 3, 1879.

BOHNALITE

The Latest Process Light ALLOY

Bohnalite, the latest process light alloy is playing an important part in aircraft motor construction. Its great lightness and great strength are vital features in aircraft pistons, cylinder heads, crankcases, etc.

As we are constantly experimenting with the foremost aeronautical engineers, it is possible that Bohn engineers and Bohn metallurgists may be of service to you in connection with some of your motor problems.

Investigate Bohnalite—the advanced light alloy.

BOHN ALUMINUM & BRASS CORP., DETROIT, MICH.
Also makers of the famous Bohn Ring Valve Bearings



TRADE MARK FOR REGISTERED AVIATION

Fairchild

FIRST Again...

GREENLY ISLAND

410 MILES

ICE FIELDS

SEVEN ISLANDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS

ICE FIELDS



The crew in search of the disabled Beech—on the Fairchild Cabin Monoplane

THE first plane to reach the Beech's crew at Greenly Island was a Fairchild Cabin Monoplane. No accident was the dimmest of the disabled Beech known, then C. A. (Duke) Schiller and Dr. Louis Cousinier of the Transcontinental Airways, Ltd. of Canada, volunteered to carry aid to the three trapped men. Beron Gustaf von Hannefeld, Major Fitzmaurice and Captain Hermann Koehl.

Through almost a thousand miles of rain, snow, sleet and fog, the Fairchild rescue plane plied its course from Murray Bay to Greenly Island. No more hazardous or unfavorable a flight could be contemplated. Into the very jaws of disaster, many eb servers predicted. The Fairchild Cabin Monoplane, nevertheless, went through, and brought Major

Fitzmaurice back to Murray Bay for spare parts and special fuel for the Beech.

Of the eight planes which subsequently flew to Greenly Island, seven were Fairchild Cabin Monoplanes. The respect with which experienced pilots consider Fairchild planes is typified by Duke Schiller's remark, when a Fairchild was suggested as the plane for his epic flight: "With that baby, we can go to the pole."

As a result of the experience gained in building more cabin monoplanes than any other manufacturer—all Fairchild planes are ruggedly constructed for dependable service over land, water, or ice—through sleet, rain and snow as well as in the hot and arid of the torrid zone, Fairchild Aircraft Manufacturing Corporation, 270 West 38th Street, New York City.

SUBSIDIARY OF FAIRCHILD AVIATION CORPORATION OF

FAIRCHILD



THANK YOU for reading AVIATION





Builders of modern machines
task mastered by expert opera-
tors not in making possible the
completion of a plane a day.

MAN~POWER No airplane builder can collect over-night or in a few hasty weeks the staff of adequately experienced engineers, designers and craftsmen required for errorless production. The Glenn L. Martin organization, numbering 750 men, is the result of eighteen years' search for and selection of the individual specialists best fitted for each particular responsibility.

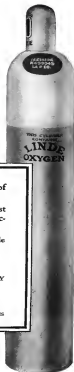
THE GLENN L. MARTIN COMPANY
Builders of Quality Aircraft since 1909
CLEVELAND, OHIO



THANK YOU for membership AVIATION



You'll buy
**LINDE
OXYGEN**



When you think of oxygen in terms of what it will do

When you realize that the cheapest oxygen is that which is used most effectively

When you know the value of Linde Process Service

Then you'll buy Linde Oxygen.

THE LINDE AIR PRODUCTS COMPANY
Unit of Union Carbide and Carbon Corporation



*General Office: Carbide and Carbon Building
30 East 42d St., New York*

45 PLANTS 365 WAREHOUSES

THANK YOU for membership AVIATION



Adaptability of the Ford tri-motored, all-metal transport

THE Ford tri-motored transport monoplane is adaptable to almost any special requirement an air-line operator may have.

It may be used as a seaplane, with twin floats. It can be equipped with skis for work in snow and ice. For work from fields at high altitudes it is equipped to meet these conditions.

It can be used for passengers exclusively—or passengers, express and mail. Its capacity is great enough to permit carrying sufficient passengers on short trips to make the work profitable. On the other hand, if the run is a long one, it can carry sufficient fuel for a non-stop flight, and still leave ample space and lift for carrying pay loads.

In short, whatever the route, the location, the type of work an operator is carrying on or contemplating, he will find a study of the capabilities of the Ford tri-motored, all-metal transport monoplane—as applied to his own work—useful to him in forming and carrying out his plans.

We will be glad to consult and advise with air-line operators. This service is rendered without cost or obligation to those requesting it. Communicate direct with

THE STOUT METAL AIRPLANE COMPANY
Division of
FORD MOTOR COMPANY
Dearborn, Michigan

TRADE TALK Air Line/AVIATION



The Oldest American Aeronautical Magazine

Vol. XXIV

MAY 21, 1928

No. 21

Too Many Ideas

IN THE early period of development of almost any industry, those who have the courage to do the pioneering work have a strong desire to develop their own ideas. This is in itself an excellent characteristic, especially if the ideas are sound, but also it can be carried too far. There must be a certain number of people who remain impartial and judge whether the ideas are sound or not and who do not back sound development by trying to impose ideas of their own which they wish to see developed.

In the commercial field the buyer is the ultimate judge and if a product or an idea is not basically correct it gradually fades out of the picture. In the military field the check is much more difficult. The top power who is the ultimate user is too far removed to have any except the most general control. Even the flyer who is the actual user in several steps removed from the actual purchasing power and even he cannot reduce his experience to the easy standard of whether the equipment pays for itself. Between the flyer and Congress, who makes the purchases, come several intermediate steps all of whom have their ideas and as a result impartial testing is hard to get.

There is perhaps no field in which this is more true than that of instruments. Much of the work must be done in the laboratory but the testing must practically all be done in actual flight. In aviation work as well as in the military field it is extremely hard to get an impartial test of instruments. Not only have the engineers in charge of the test work get their own ideas but they get a pretty good notion of what an outfit can do. When a new instrument is brought up the maker has the approved difficulty in getting an impartial and thorough study of its possibilities and defects. The difficulty lies in a considerable extent in the fact that those doing the testing have too many ideas of their own.

Cheap Cigars

FOUR ALL appearances these seem to be quite a number of persons who are interested in aeroplanes and in some cases are actually making their living out of the industry, who do not make use of the air machine for their usual important business. Undoubtedly one reason for this is the fact that they have been careless or lazy about obtaining air mail stamps and seeing to it that the stenographer forms puts them on the proper items.

This office has adopted a very simple (and effective)

method of making sure that the air mail is used. Orders have gone forth that all destined letters to points on an air mail route, or more than a night's train journey away shall go by air mail. If the stenographer forgets to use the air mail stamp the writer (by checking up when he signs the letter) has the right to refuse one signed from the stenographer. True, the quality of the paper has not been stipulated, but there is a certain satisfaction in receiving a free cigar even if it is not a very good one.

The Connecting Link

ONE YEAR ago Col. Charles A. Lindbergh spanned the Atlantic and linked New York and Paris by a non-stop flight. The feat was marvelous enough in itself but its results have been even more extraordinary, because that flight, which in many ways had the aspects of a stunt, has proved to be the connecting link between commercial aviation and the public.

As a matter of fact real commercial aviation existed in this country before the epic events of last spring. However, the public did not realize it, and its possibilities lay dormant until Lindbergh's feat and his extraordinary personality brought aviation to its attention. Airplanes are little different today than they were a year ago, but the industry is a far different thing than it was when Lindbergh made his great flight.

The effects of that history-making flight upon the growth of the aeronautical industry have been very noticeable as spontaneous as the flight itself. During the year that has passed much Lindbergh thundered down the Roosevelt Road, wherever the enthusiasm for them pertaining to aeronautics has spread like wildfire throughout the entire civilized world. Hundreds of landing fields have been constructed and plans and preparations have been completed for the establishment of thousands more in the very near future. Aeronautical enterprises which were absolutely unable to finance themselves before the great New York to Paris flight are now able to secure the necessary capital with little or no investigation of the merits of the proposition. Airplane and airline theorists have sprung up like mushrooms all over the country and all are anxious to do the orders that have come piling in. Flying schools and taxi services that were once just barely getting by are now doing a great volume of business. In short, the aeronautical industry has at last come into its own, and the old timers who have watched so much that aviation might live are celebrating the opening of this connecting link and the realization of their fondest dream.

The Curtiss "Robin"

New Commercial Cabin Monoplane Powered with an OX-5 has a High Speed of 99.7 M.P.H. and Lands at 40 M.P.H.

By ROBERT E. OSBORN

THE NEW Curtiss "Robin" has made its appearance, quite appropriately, in the springtime, and, like its predecessor, promises to become more numerous in the fields in the summer season. It was unfortunate that flight tests on the new plane were not completed in time for it to be exhibited at the All-American Aircraft Show at Detroit, its judging by the interest shown by the arrangement of the new plane and the literature describing it, the exhibit should have been a very popular one.

So far it has been flown extensively by "Crazy" Jones and Paul Boyd, both of the Curtiss organization, M. M. Merrill, manager of the Curtiss Flying Service, Col. Clinton A. Lindbergh, Maj. W. R. Robertson and several of the flying service pilots. The immediate reaction of most of those who have flown in it, or have even seen it, seems to be to ask Major Robertson, of the Curtiss-Robertson Airplane Manufacturing Co., for the agency in some particular district. Major Robertson says he has enough requests for the agency so far that it seems one could be placed in every county in the United States, but that no contracts have been signed, or will be signed, until the construction program is completely organized and the plane actually on a construction basis, so that delivery can be made and kept. Everyone is well pleased with the plane, and its performance, appearance and general controllability and maneuverability, these many compliments from all who have flown in it.

The Robin is a high wing monoplane with the airfoil section maintained across the top of the fuselage. The cabin is mounted in this wing and at the top of the wing forms the top of the cabin, the depth of the nose gives additional headroom. The lift train is composed of two struts from two beams of the cabin to a small auxiliary strut under the wing. These two main struts are each enclosed by ribs and a covering which forms the main airfoil section at the nose.



First quarter view of the new Curtiss "Robin" (high wing) monoplane in flight.

wing. These strut wings extend only to the auxiliary but as they would interfere with the main wing beyond its point. The power plant is the 60 hp. Curtiss OX-5 engine. The Curtiss-Robertson Company is in the fortunate position of having 1950 of these engines and a large store of spare parts. The plane has also been analyzed structurally to be



Side view of "Robin" in the air.

the new Curtiss "Challenger" six-cylinder engine which develops 150 hp. The Robin uses a pilot and two passengers.

The aircraft are the usual unbalanced type, which has been found to give splendid lateral control. The roller of stabilizer are not balanced but give full control at the end with very low stick loads over the flying range. The cabin is readily adjusted by a lever at the pilot's left side.

By raising the wing lift train as part of the landing gear train, a light wide tread landing gear is obtained. It

May 21, 1938

was arranged that the main struts between struts, which are necessary to meet landing gear arrangements, and which cause extra weight and aerodynamic interference, are omitted. The struts are built about the pilot to handle any place on the ground as well as any place equipped with dual brakes.

The plane was completely designed, and the first three or four are being built by The Curtiss Aeroplane and Motor Co., Inc., in Garden City, Long Island, N. Y. Production on



Colonel Lindbergh and Major Robertson in front of the new Curtiss "Robin".

a large scale will be taken in the newly constructed factory of the Curtiss-Robertson Airplane Manufacturing Co., at Garden, near St. Louis, Mo. This factory, which is now completed, has been built now for the production of this airplane only, and therefore production should be smooth and its planes turned out in satisfactory quantities.

Models of the airplane were tested in the Curtiss wind tunnel for aerodynamic qualities. The high performance of the plane is directly traceable to these tests as numerous wing and cable arrangements were tested and the best was selected. Adequate control in any attitude of the plane was also secured by these tests.

Structural analysis was complete and the structure at every point has an additional margin of safety over the requirements of the Department of Commerce. The long experience of the Curtiss engineers in building their designs in service conditions enabled them to make additional allowance for parts of the structure which are likely to get hard service, or for materials which tend to deteriorate with age and weathering.

All materials are tested by sample, and numerous drop tests were made on the landing gear and tail wheel, and complete load tests were made on several surfaces, ribs, beam and numerous fittings. The assurance is rather well established that it is an aerodynamically and structurally safe airplane.

Following is a summary of the performance tests of the Curtiss flight test section. These trials took two weeks to complete and were run in exactly the same manner that is now done for the Army or Navy is tested. All data is checked in standard strengths.

Flight Test Summary
Robin Cabin Monoplane, OX-5 Engine
(All data below is for the full load condition)
Empty weight 1,175 lb.
Landing speed 48.8 m.p.h.
Empty ft. 69.7 m.p.h.
High speed (low level) 99.7 m.p.h.
Empty ft. 91.8 m.p.h.
Service ceiling 12,750 ft.

AVIATION

1451

Absolute ceiling 16,000 ft.
Take-off time, full load 7 to 9 sec.
Climb to 5,000 ft. 1.4 min.
34,000 ft. 3.6 min.
Service ceiling 12,750 ft.
Climb to 10,000 ft. 5.76 min.

Wing loading 16.50 lb./sq. ft.
Endurance, full throttle 2.5 hr.
Range, full throttle 375 mi.
Endurance at 80% max. rpm 4.2 hr.
Range at 80% max. rpm 400 mi.

General Characteristics—
Wing loading 16.50 lb./sq. ft.
Airfoil section Curtiss-72
Landing gear, high incidence
85 Dept. of Commerce requirement
G.O. in per cent. of max. aerodynamic chord 23.5
Fuel consumption, full throttle 49.0 lb./hr.
Oil consumption, full throttle 3.9 lb./hr.
Fuel consumption at 80% max. rpm 12.2 lb./hr.
Oil consumption at 80% max. rpm 2.0 lb./hr.
Fuel capacity 30 U. S. gal. total, 30 gal. normal load
Oil capacity 4 U. S. gal.

Dimensions—
Length, overall 28 ft., 6 in.
Height, overall 7 ft., 4 in.
Span, overall 39 ft., 6 in.
Chord 12 ft., 2 in.
Wing area 16.50 sq. ft.
Incidence 3 deg.
Dihedral 1 deg.
Sweepback 0 deg.

Areas—
Wing, including ailerons 265 sq. ft.
Ailerons, total 22 sq. ft.
Horizontal tail area, total 36.3 sq. ft.



Typical one-plane strapped "Robin" with wing truss in emergency landing. Weight of rib 48 of a pound. Load below, 125 lb. without jacking.

Vertical tail area, total 152 sq. ft.
Elevation, total 14.8 sq. ft.
Rudder 9.8 sq. ft.
Among the unusual structural features of the new plane, an outstanding one is that the wing ribs are made entirely of the new Alclad metal (aluminum with a pure aluminum coat). Each rib is pressed into shape from a single sheet. These ribs weigh 30 lb. each and take a properly distributed load along the chord of 535 lb. without yielding. These ribs are placed on spaces equal having the normal type of drag brace.

Continued on page 1452

The Hallett Aircraft Engine

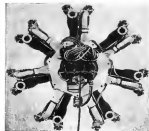
By CHARLES F. M. REYNOLDS



Front view of the 120 hp Hallett engine.

WITH THE first test of 56 engines conducted for and now in production for delivery within 90 days, the Hallett Manufacturing Co., Los Angeles, Calif., is definitely in the market with a new radial aircraft engine. The Hallett model H-120 is a seven-cylinder, single-cylinder, air-cooled engine rated at 120 hp when turning a maximum of 1800 r.p.m. The cruising speed is 2500 r.p.m., displacement is 528 cu. in., total weight 420 lb., bore 4 1/2 in. and stroke 5 in. After years of study and experiment the seven-cylinder 120 hp. model was developed because of its simplest, sturdy, new and the fact that a 120 hp. engine will probably be the lightest equipped for many of the commercial airplanes now being built.

Apparently overvalued in design and construction, the Hallett engine is most noteworthy for the extensive engineering work which has been done with a view to making it an ideal commercial engine as adapted to automatic production methods that it may be built and sold at an exceptionally low price. So carefully has the air-cooling of this engine been worked out that the highest temperature reached during the



Low maintenance view of the Hallett H-120 engine showing the Zenth magneto and the Zenth carburetor.

test run was 350 deg. just below the exhaust port. The maximum variation of temperature in the cylinder head is said to be under 30 deg., which is exceptionally low for an air-cooled engine.

Availability of an ideal test is also provided. To disassemble the engine it is only necessary to loosen one flexible hose on the outside and the mounting bolts to remove it from the engine housing. The engine may be disassembled from the front of the airplane without removing it from the fuselage mounting. The only tools necessary are the propeller hub or oil pump 3/4 in. and a 7/16 in. open end wrench. The engine may be completely disassembled without pulling it apart. All accessories are mounted on the open end and may be reached without disassembling any part of the engine.

Realizing that in commercial work a product must stand against abuse, Mr. Hallett has built the engine very sturdy, and it is expected to prove of exceptional reliability and durability. No appreciable wear has been shown after testing this engine for hundreds of hours on the test stand. The comparatively high weight power ratio due to this substantial construction is offset by the low fuel and oil consumption. The Hallett engine operates at full load on less than one gallon of gasoline and half pint of oil per hour.

Low Maintenance Costs Possible

Simplified design and machine production have made possible not only the extremely low first cost, but assurance of low maintenance costs. Interchangeable parts and simplicity will lower down overhead and total cost born on overhead jobs, it is believed.

Two Zenth carburetors are mounted at the rear of the engine and dual ignition is provided with two Zenth magnetos. The ignition system is forced feed with a scavenging system which includes air-cooled pump and scavenging pump. Dual systems is provided using two Zenth magnetos.

Each of the cylinders is cast in one piece with integral fan. The cylinders are of special composition grey iron extensively machined with boring, facing and honing being performed in one operation. Four long and two short Mac Wipac tie rods hold the head and cylinder in the crankcase.

Continued on page 1453

Speeding Up Production in the Factory

By WILLIS PARKER

0 WISE TO the demands of the industry, aircraft manufacturers have had to devise many appliances, peculiar to their needs, in order to speed up production. These appliances have been constructed by the Alexander Aircraft Co. and are used in connection with gluing. One of these appliances is the gluing of the long boards from which the wing ribs are made another speed up the gluing of and lots of wood used in the construction of the wing ribs. The third one down is a jig, the time it takes to alter in length of tape placed over the areas and where ribs of ribs fasten the ribs covering to the wing frame.

Because of the length of the wing spans—up to 35 ft.—the process was not available. Many manufacturers have two and still are using "C" clamps to hold the boards to, after while the glue dries. D. M. Alexander, chief of the engineering department of the Alexander Company devised a very simple press, in which 25 in. boards, or 12 ft. span, may be pressed at one time.

A row of slatbars were built with a pressure plate at the bottom coupled with a pressure plate at the top, against which the ribs, spaced 15 in. apart, the full length of the frame—25 ft.—press under a patented pressure of two tons each.

When 12 pairs of boards have been placed, they are placed in the overhead, and a long 3/4 in. steel beam is placed on top of them. The jack screws rest on top of the beam with their heads against the upper pressure plate which, as previously mentioned, is coupled to the bottom pressure plate by steel rods. The pressure is applied by turning the screws with a small bar.

Since the steel beam is the same width as the boards—eight inches—the pressure on the boards is equal over the entire surface. The jacks spaced 15 in. apart because equal pressure the full length of the press.

To facilitate removing the boards when they have thoroughly dried, arrangements have been made to loosen the jacks. In the steel beam two inches, insert a bar in holes in the steel on frame and then hold the bar and the jacks up. The steel beam of spurs is easily for pressure. By this action the spurs may be loosened at one time at 24 in. The bar required to clamp them down has been ordered at 100 lb. per cent. over the old system of clamping each pair of boards with a series of C-clamps.

In the construction of the wing ribs, numerous thin pieces of lumber must be glued into narrow grooves in other strips of wood. To apply the glue with a brush would take considerable time. So a device that might be likened to a paint tray was constructed from an oil can and a pair of nail-making iron tubes. The bottom of the oil can was removed and the inner tube fitted over the can, with a clamp. The speed of the one movement, leaving an opening large enough to pour in a quantity of glue. Then the top or spool is re-



A view of a machine in which the 12 ft. glue press developed by the Alexander Company.

turned, and by applying a little pressure in the inner tube, the liquid glue is forced out of the spout in a small stream which may be directed along the groove in the wood by the worker. The bottom end was sealed, of course.

One girl, equipped with this little arrangement, can apply the glue to a series of grooves just as quickly as she can slide the tip of the oil can from one side of the frame to the other.

Over on the wing end panel department a similar device allows glue to be connected with placing the strips of tape over the wing frame to protect the ribbing. The tape is usually mounted in the shape that is used on the wing fabric to tighten the cloth and usually is applied after the first coat of dope—dope—has been applied to the cloth. The most method of applying the tape, is to cut off a piece the required length, dip it in a bucket of dope and then slip it over the corner. This takes too long for the Alexander people, so they devised a five gallon tank on roller platforms. On top of the tank is a spool for the roll of tape, which is unwound from the spool as needed, goes in a continuous strip down to the bottom of the tank, around a spool at the bottom and in to the top again and through a slit in the cover. The tank is filled with dope. Obviously the tape, passing from the upper spool, into the tank, around the lower spool and up to the top again is thoroughly saturated with the dope. Two men work the appliance. One is on the opposite side of the wing and takes the end of the wet tape from the machine and lays it back with the same on his side. His assistant pulls off the length with a pair of scissors and their smooth it down in its proper place. Since the tank is on rollers, it can be wheeled to any position of the room desired and can be moved along the length of the wing in laying process.

The Stearman Biplane

Models C-2B and C-2K were Designed for All-Around Service and Differ Only in Engine Installation

THE STEARMAN Aircraft Co., recently of Vanis, Calif., and now located at Wichita, Kan., is in production on a light commercial or private owner airplane. The plane was originally designed and engineered before construction, rather than after production had been started and as a result, few changes have been necessary. This fact is well exemplified by the service that Stearman mail planes are going to the air mail agencies of the west and southwest.

In the conception of the design, an effort was made to produce a non-volatile variable airplane. The result is that one basic design has been created that will meet the needs of the private owner, the flying school, the commercial operator, or the air mail contractor. Consequently, the type designed was adapted to various power plants. The range which was selected for the power plant installation was from 90 to 230 hp. This permitted an increase in power of substantially three fold without changing the structure at all of the air mail. In order that the craft would be adaptable to production with these various power plants, it was deemed advisable not to design to a particular engine but to the contemplated maximum power. The load factors which were to accompany this condition appeared to require somewhat of a handicap in the structure and performance when engines of lower power than the maximum were used. However, as a result of an investigation of the problem of variation in weight and performance with increase in load factor, the engineering department of the company found that the rate of change of weight with load factor would be approximately one per cent. That is to say, the structure would increase in weight one per cent per unit increase in load factor. The change in performance of the lower powered models due to the increase in load factor was not appreciable. It was therefore deemed entirely justifiable to design for the maximum power. A high stan-

dards load factor value of 8.5 was chosen, with the other dimensions proportional to this figure.

The structural advantages of a small lower wing and its aerodynamic superiority of a highly staggered, high upper wing arrangement is manifest in this design. The upper wing is constructed of two panels and a center section, while the lower wing is constructed of two panels, one on each



An upper wing panel. Note the dorsal leading edge and the internal support bracing.

side of the fuselage. Spruce is the dominant material used in the construction of the wings. The beams are of round I section, while the ribs are of truss construction. The leading edge is covered with duralumin sheet for the entire span. The major portion of the wing stringers are fabricated from heat treated duralumin sheets and are treated with aluminum varnish. Production of these parts has been planned so that three different shaped fittings and wire lugs were for all the attachment points. Forward drag brace, serving as a tie for the rods, while the external wing bracing is of aluminum tie rods and struts. Each of the outer panel

Continued on page 1479



Front quarter view of the Stearman biplane, model C-2B (230 hp. Ryan-Siemens engine).

Fuselage Analysis

Stress Analysis of Commercial Aircraft, Chapter Number Eleven

By PROFESSOR ALEXANDER KLEMIN

David Greenglass School of Aeronautics

And GEORGE F. TITERTON

Staff of the Bureau of Aeronautics, Navy Department

THE FUSELAGES of American commercial planes are almost all welded steel tubular structures. The steel may be either mild carbon (1020) or chrome-nickel-phosphorus steel. There are a few planes with dural fuselages, and some fuselages are still constructed of wood and wire. Free-layers of dural are probably somewhat ahead of normal development but will become more common as the question of attachment becomes better understood. The covering of fuselage will be of welded chrome-nickel-phosphorus steel, except for the engine mount which will be of mild carbon. Using mild carbon for the engine mount gives tubes which are greater in diameter and somewhat easier to weld than the chrome-nickel-phosphorus tubes, and are more suitable for use where engine vibration is involved.

In the design of the fuselage it is of course necessary to place panel joints at the front of the wing and choose attachment points, and to have a sufficiently deep tail post to

house the tail surfaces. In addition the tail end fuselage must be well braced. Structural members likely to impede the vision of the pilot should be as few and as small in diameter as it is also desirable to have the tubes entirely free of branching members.

The tubular members employed in the fuselage trim are of course, designed as minimum-weight long or short depending upon the material, length, and diameter required. The efficiency of fabric, G, to be used in the design of these members depends upon the method of fastening employed. For welded joints a coefficient of $G = 3$ is allowable; for well gusseted riveted joints $G = 1\frac{1}{2}$ is safe; and for pin-ended stress $G = 1$ must be used.

Requirements of the Commerce Department

In Chapter 3 of the requirements for the fuselage analysis were explained generally. They will be reviewed here in detail.

Continued on page 1472

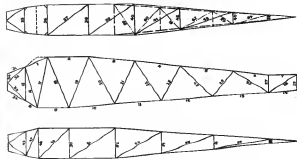


Photo Machine Drawing (Revised), showing the fuselage structure as shown in the drawing.

LINE DIAGRAM OF FUSELAGE, Fig. 11

Etching of Aluminum Alloy Propellers

BY A. J. LYON
DETROIT, MICH.

A LUMINUM ALLOY propellers are subjected to a light etch, both in the process of inspection and periodically when in use, for the purpose of detecting cracks or other defects that may be the cause of failure. The light etch, which should be distinguished from the deep etch used to reveal the structural features of the material, is accomplished by etching the surface of the metal to the extent of a coarse sand surface for a short period of time, removing it in water, and then removing the deposit of copper which forms during the etching operation with a water solution of nitric acid, and finally washing and drying. This is followed by a careful inspection and all apparent defects are marked. The small areas thus marked are then deep-etched locally for the purpose of definitely identifying the apparent defects.

Tests on specimens for the caustic solution should be constructed of steel or wood and the size determined by the type and number of propellers etched. The manufacturers of detachable-blade propellers have found that cylindrical tests inscribed in a vertical position, made of either steel or wood, are satisfactory. A rope or chain hoist and an overhead trolley are convenient for handling the propellers. At the Air Corps stations where all types of propellers must be etched, it is found that tanks approximately 12 ft. in length, 18 to 24 ft. high, and 12 to 18 in. wide, installed in a horizontal position, are satisfactory. Provision for heating by means of oil or steam or gas should be made on all permanent installations. The acid tank should be similar to the caustic tank in design and construction, but steel cannot be used for this purpose. Wood or welded aluminum sheet has proved to be satisfactory. Likewise, the tanks or vats for the rinsing water should be practically the same size as the acid and caustic tanks. It is preferable to have two, one equipped to heat the water for the purpose of preventing the propellers and for the final washing, and the other for use in rinsing the excess caustic and acid from the surface of the etched propeller.

A 10 to 20 per cent. water solution of commercial caustic soda is used. In mixing simply add one to two pounds of caustic soda to each gallon of water, care being observed not to inhale the fumes or to allow the solution to splash on the clothes or hands. The concentration can be checked by measuring the specific gravity with a hydrometer or Baumé scale for liquids heavier than water. The specific gravities and Baumé readings corresponding to 10 and 20 per cent. solutions are 1.115 and 1.250, and 14.5° and 27.0°, respectively. The caustic solution may be used for a long period of time but sodium flake caustic is added the strength and concentration will gradually decrease. The specific gravity is used as an indication of the concentration of the caustic solution after it has been used for etching aluminum alloys due to the presence of sodium silicate. The concentration may be checked, however, by etching in an alcoholic solution as follows: Mix carefully one ounce of the caustic solution with

mostly two ounces of denatured ethyl alcohol and add 5 to 10 drops of an alcohol solution of phosphoric acid, which colors the alcohol-caustic solution pink. To find out a 5 per cent. water solution of chemically-pure commercial nitric acid (2 parts nitric acid to 33 parts water), or, more at a time, until the color changes from pink to a milky white. The addition of one ounce of the acid solution is approximately the equivalent to a 5 per cent. caustic solution, three ounces to a 10 per cent. solution, three ounces to a 15 per cent. solution, and four ounces to a 20 per cent. solution.

A 5 per cent. solution of nitric acid can be prepared by mixing one part of commercial nitric acid to 19 parts of water. The specific gravity and Baumé reading are 1.02

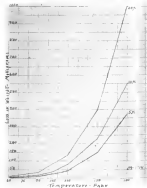


Fig. 1—Graph showing the effect of temperature of caustic solution on the depth of etch of samples of an aluminum alloy propeller. Time of immersion, five minutes.

of 100, respectively. Care should be observed in pouring in and into the water.

The methods of etching outlined herein have been used with good results and should be used as a guide. The details of the operations may be varied to meet the methods of man-

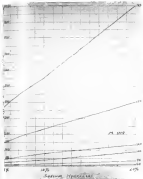


Fig. 2—Graph showing the effect of the concentration of the caustic solution on various temperatures on the depth of etch of samples of an aluminum alloy propeller. Time of immersion, five minutes.

ufacture or operating conditions and usually as good results obtained.

When propellers have been rag or color buffed in the process of manufacture an amorphous film is formed over the finish of the metal which soon defects and is more resistant to the dissolving action of the caustic soda than a surface which has been polished with a fine abrasive. The amorphous film when formed must be removed in the etching bath in order that a proper inspection can be made. Therefore, the rag buff should not be used as a final finishing operation. Good practice for etching and inspection and one that presents a good appearance is produced by simply polishing with a extremely fine abrasive, such as 150-grit emery and paste. The grease from the polishing operation must be removed by means of suitable solvents, and the same applies to oil, varnish, or other protective coatings on propellers of this type. The hubs must be removed from detachable-blade type, and Currier-type or twisted propellers. The twisted, "W" or forged-type propeller may be etched with the hub or steel lines in place provided it is properly oriented in the caustic solution as described hereafter.

The treatment of the propeller should be approximately 15 min. in each of the caustic solution. This is necessary in order to insure a uniform etch on all parts of the propeller

through the initial temperature be below that of the caustic solution, the time necessary over the tip and edge of the propeller blades will etch much more rapidly than the heavy sections at the center of the blade and around the hub due to the fact that the latter section require a considerable time interval to reach the temperature of the caustic solution due to the mass of metal involved. The propellers can be preheated by immersion in hot water and allowing them to remain for a sufficient length of time for all parts to reach a temperature slightly above or equivalent to that of the caustic solution. Water near the boiling temperature can be used to advantage for this operation. Before etching, the bearing surfaces on the hub centers of the detachable-blade propellers and the steel hub of the Currier-Ried "B" type propeller should be covered with a grease having a relatively high softening point. The grease may be warmed and then applied with a two or three inch paint brush or comb.

The propellers are then immersed in the caustic solution. The depth of etch is most accurately regulated by the length of time the caustic solution is allowed to remain in contact with the metal surface. Methods of determining the proper length of time required for a satisfactory etch are to judge the color of the copper deposit, to measure the difference in thickness before and after etching, and to observe the etched structure. Propellers in the process of manufacture should remain in contact with the caustic solution until a decrease in thickness ranging from 0.005 to 0.007 in. has been obtained. The time required can be determined very accurately, if the temperature and concentration of the caustic solution are held constant, by using small samples of a propeller alloy of known area and weighing before and after



Fig. 3 (left)—Photograph of etched cross-section of the hub section of a detachable-blade propeller. Arrow points to crack formed during service of the propeller. Fig. 4—Photograph of the tip of an aluminum alloy propeller, deep-etched. Arrow points to hub forward direction (top).

etching, or by measuring with microscopes, care being observed to make all measurements at the same temperature. Propellers that are in service and have been previously subjected to an etch require a very short etching period and the amount of metal removed should not be measured. The depth of etch has been obtained by measuring the depth, for periods of time ranging from two to six minutes. Where the surface was polished and color-buffed they require five minutes, and where the surface was simply polished, two and one-half minutes, to produce the desired depth of etch. With a 20 per cent. solution at 175° F., polished propellers were

Continued on page 1457

Storms Delay Looming Amphibian Flight To Retrieve Bremen from Greenely Island

STORMS ARE delaying the retrieval of the trans-Atlantic Junkers monoplane Bremen, at this writing. Maj. Gen James E. Fehrer, chief of the Army Air Corps and commander of the two Looming Amphibian rescue planes, is at Fort M. S., with his pilot, Lt. Col. Edward R. Quenda, awaiting better flying conditions for the construction of the trip to Greenely Island. The second amphibian, piloted by Capt. Ira C. Baker and bearing Fred Melcher, Junkers pilot who is to bring the Bremen to New York, has been forced down in a snow storm some 50 mi. short of Folsom.

The two Looming Amphibians left Walling Field, Washington, on May 21 and were down to Miller Field, Staten Island, N. Y. On May 22 the journey was continued to Furland, Me., and St. John, N. F., after Fred Melcher had been picked up at Miller Field. Lt. Col. Quenda, the first Looming, near St. Fowble had picked the second plane, but he was forced to give his place to Lieutenant Quenda at St. John because of falling ill with appendicitis. The flight was again postponed from St. John on May 14, only to be stopped by adverse weather at the Folsom vicinity.

If landing conditions are good on the established trail near Greenely Island, to which the Bremen has been removed, Melcher will drop by parachute. General Fehrer's party in the Looming planes will await the Junkers pilot. If he is successful in getting the Bremen into the air, the three planes will fly back together immediately, but if trouble is encountered in removing the ill-stricken monoplane, then other arrangements will be made.

Government Officials Confer With New York State Committee on Promotion of Aviation

A CONFERENCE on aviation between officials of the Departments of War, Commerce, and the Navy and a New York State legislative committee is being arranged, at this writing, to be held on May 19. At that time, the four assistant secretaries in charge of Government aviation will confer with the New York group in the office of Assistant Secretary of War F. Trotter Deverson in Washington. Among those attending will be Assistant Secretary of Commerce William P. MacGowan, Jr., Assistant Secretary of the Navy Edward P. Warner, and Assistant Postmaster General Warren I. Glavin.

The Joint Legislative Committee, which is the first of its kind in this country, will make a study of Federal and state government promotion of commercial aviation. Relation of the state airports to national defense will also be examined. The New York committee, headed by State Senator J. Gus Walsh, whose name was named by New York's recent aviation bill, will fly to Washington for the conference, it is said. A Curtiss plane, piloted by "Candy" Irwin of the Curtiss company, and a Fairchild cabin plane, with Lieutenant Barnes at the controls, will carry the group. Members of the committee are Senators A. Spencer Field and Henry J. W. Ladd, and Assemblymen Gordon G. Parsons, Herbert B. Smith, William H. Tugent, and Andrew B. Hyattson.

Texas Air Transport Co. Beginning Regular Dallas to Houston and San Antonio Service

REGULAR AIRCRAFT service is being inaugurated by the Texas Air Transport Co. over two lines from Dallas, Tex., to Houston and San Antonio, Tom Hadden, chief pilot, announced. Two Ryan cabin monoplane have been ordered for the service, and one of them has already been delivered from the factory at San Diego, Calif., by Mr. Har-

den. Additional passengers will be carried in Pitcairn Mail wing planes, it is announced. The Texas Air Transport Co., which is contractor for two air mail routes, maintains headquarters at Fort Worth and keeps field representatives along the air mail route in Dallas, Waco, Houston, Galveston, San Antonio, and Austin. The company also operates a flying school at the new Houston airport.

State Air Tour Conducted in Oklahoma As the Cities Observe "Aviation Week"

GOV. HENRY S. JACKSON set aside the week of May 18 as "Aviation Week" for Oklahoma. During the period the Oklahoma State Air Tour was being conducted, and La. Mail Week was also being observed. In Oklahoma Gov. Henry S. Jackson distributed 20,000 letters containing air schedules, and special letters were sent to heavy business postage urging them to use the air mail.

Other visits by the state air tour were Oklahoma City, the starting point, El Reno, Guthrie, Ponca City, Red, Stillwater, Chickasha, Norman, Duncan, Ardmore, Durant, Weatherford, Muskogee, Broken Arrow, Muskogee, Miami, Bartlesville, and Tulsa. More than 25 planes were entered.

The tour was headed by the Oklahoma State Chamber of Commerce, which covered many airports for cities by the air tour and also conducted much interest for commercial aviation. The Texas Company's air-touring Ford, two Ryan biplanes, two Stearns cabin planes, three Sperry, two Ryan, Engineering, Waco, and American Eagles were plane entered in the tour.

No unnecessary restrictions or measures were permitted in the tour. No prizes were offered of any sort as the tour was not a contest. M. H. Ketchum, Oklahoma National Association Aviation governor, was commander, Col. E. H. Haggard, president of the State Chamber of Commerce was manager, and Billy Burke of Okmulgee was flight leader.

The general aviation committee of the State Chamber of Commerce arranged the tour. B. H. Barry of Tulsa was chairman, while E. A. McElroy, Tulsa, Oklahoma, Glen F. Baker, C. C. Roberts, R. C. Martin, A. E. Warner, Daniel T. Billy Parker, and H. C. McElroy were members of the committee.

Several months preparation were made before the tour, and the success of the tour was the result, as well as great good will feeling toward aviation among the people of the state.

A Radically Different Monoplane



Fairchild's modern wing monoplane, designed by Martin Bell of the Dues Aircraft Corp., Detroit, and recently sold to Mr. Gorman, Mich. The wing area is now being prepared and an Anzani engine is being placed in the craft. It will soon start showing the plane's construction appeared in Aviation for May 7, 1928.



Especially for Private Owners THE PITCAIRN SPORT MAILWING

In the Pitcairn Sport Mailwing there is offered a quality and degree of performance that accords it a unique position. It is a sport ship with all service characteristics of the finest types of planes used in the air mail service, and as such, appeals to the most discriminating class of owner-pilots.

It is a companion ship to the familiar Pitcairn Mailwing*, only differing by the conversion of the mail compartment to an additional cockpit for passenger-carrying. There is also added a baggage compartment directly behind the pilot's seat.

The Pitcairn Sport Mailwing answers the demand of private owners for a plane of wide speed range, remarkable stability, and of the highest maneuverability. It is powered with the Wright Whirlwind J-5-G engine and furnished with the following instruments and equipment:—

Instruments	Equipment
Compass	Altimeter
	Navigation Lights
	(Also wired for landing light)
Tachometer	Gas Gauge
Oil Pressure Gauge and Thermometer	Fire Retard Mechanism
Clock	Five Extinction
	Electric Fuel
	First Aid Kit

Price, fully equipped at Bryn Athyn, Pa., \$9,850. Send for illustrated literature and complete specifications.

PITCAIRN AIRCRAFT INC.
1848 LAND TITLE BUILDING • PHILADELPHIA, PA.

*See U. S. Pat. Off.

Last Minute Briefs

D. D. Johnson of West Harrison, Penn., has been awarded the ownership of Luscombe-Peage plane as 15 member of Northeastern Pennsylvania. Johnson has purchased a Luscombe-Peage from the United Air Transport, Inc., of Philadelphia, Pennsylvania and New Jersey distributor.

The Adkins Aviation Co. of Gary, Ind., has been named Indiana distributor of Monogram planes. The first of an order for 10 of these planes will be turned this month.

The Federation Aeronautique Internationale has recognized the national flag of 35 ft. 38 in., 30 in., established by Edward A. Stearns and George W. Haldeman in a Stearns-Dunham plane March 28-29-30 at Jacksonville, Fla.

Officials of the Bruce Aircraft Co. of San Francisco announce that the company is again in production of planes. Two will be completed in May with an increase in cost of each succeeding month.

The Omaha-Wisconsin Airway Association, the formation of which was announced last week in AVIATION, has been informed by the postal department that an air mail line between the cities is not feasible at this time because of lack of business justifying it.

An agency for American Eagle planes in Oklahoma has been taken by the Oklahoma Air Transport Co. at Norman. The company has been in operation several years.

The United States Senate has now passed the House bill authorizing preference in air mail postage rates from ten to five cents for each mile or fraction of an ounce.

Under the organization of Roland H. Spaulding, a course in aeronautics will be offered at the New York University summer school this year. The Daniel Guggenheim Fund for the Promotion of Aeronautics has arranged for the instruction.

Feasibility of dirigible contact with ships at sea was further demonstrated recently when the Army air-craft dirigible T-30 landed as an after-dock platform of the steamer American Trader, lying off Ambrose Light, and took off a passenger and mail.

Capt. George H. Widman and Lieut. Carl E. Eason, who flew across the North Polar region, have arrived at Tromsø, Norway, aboard the Hvalfryn from Spitzbergen. Their last-landed Vega plane was flown across a six-mile ice-covered stretch from Grønsund to the Hvalfryn, upon which it was placed to be carried ashore.

The Consolidated Instrument Co. of America, Inc., New York City, accepts receipt of a contract from the Alexander Industries of Houston for 1200 Consolidated Type A aircraft instrument panels.

The Eshelby Manufacturing Corp., maker of Plymouth, Rockville, and Plymouth, recently moved to larger quarters at 120 S. La Salle Street, Chicago. The new offices are on the same floor as the Vitafine Company and other Maynard interests.

Franklin Rose and Sam Messner, Oakland, Calif., pilots, have taken the agency for Goosens biplanes, manufactured by the Stearns Aircraft Co., Wichita, Kan. Rose and Messner will be sub-dealers under Walter Varney of the Varney Air Lines.

Greiner Looming, head of the Looming Aeronautical Engineering Corp. of New York, has purchased the Dill Moth plane imported by Air Associates, Inc., of New York. The Moth is equipped with Handley-Page slotted wings.

Rolls-Royce, Ltd., of England has brought out two new engines, the F-12 and F-15, water cooled, V type engines of 19 cylinders, differing only in gear ratio. They are rated at 490 hp. and are said to develop 600 hp. The weight is 2,000 lb., while the second model weighs only 1,250 lb. The propeller speed of the F-12 is 1,425 and 1,354 for the F-15.

Scientists and aviation leaders assembled at Langley National Laboratory, Langley Field, Va., on May 15 for the third annual aircraft engineering research conference under the direction of the National Advisory Council for Aeronautics.

Pan-American Airways, Inc., reports 229 passengers and 21,600 lb. of mail carried between Key West and Havana in the month of April. Eight passenger Fokker cabin planes powered with three Whirlwind engines are used over the route.

A new system of consolidation air and rail passenger transportation across the continent is planned by a combination of companies among which are the Curtiss Aeroplane & Motor Co., the Wright Aeronautical Corp., the National Air Transport Co., the Pennsylvania Railroad, and the Adams, Express, & Santa Fe Railroad. C. M. Keys, president of the Curtiss company, leads the combination.

Van Lear Black, Baltimore publisher and aviation enthusiast, is now en route to Tokyo from London on a three-engine Fokker plane he has hired to take him to the Japanese capital and back—a trip of 43,000 mi.

The North Star Aircraft Corp. has been organized in Los Angeles with \$50,000 capital. Capt. Bruce Mather is president of the company whose headquarters are located in St. Cloud, Minn.

Grays Harbor Airways of Aberdeen, Wash., will begin production of the American monoplane as soon as the stress tests are completed. Special jigs, dies, and machinery have been gathered for the manufacture.

A new Zanders plane has been brought from Germany and assembled at Curtiss Field, L. I., N. Y. The plane is a duplicate of the Bremen, and since the cabin has not been fitted for passengers, the plane would be very suitable for a trans-Atlantic flight.

The Junkers Corp. of America now has a set of jigs and dies at Curtiss Field, enabling the manufacture there of Junkers planes.

A course in navigation and meteorology has been begun at Western College of Aeronautics, Los Angeles, with as its national southeast of 34. John C. Simpson is instructor of the new class.

A NEW TRAINING PLANE



The Curtiss Fledgling

THE FLEDGLING was constructed as the winner of a Navy design competition, in which fourteen leading aircraft manufacturers participated.

It is specifically designed for primary and advanced training, either as a land

plane or as a seaplane, and is equipped for fixed and flexible gunnery instruction, radio spotting, and bombing.

In design and performance, the Fledgling is eminently representative of 1928 standards for training planes.

The CURTISS AEROPLANE & MOTOR CO., Inc.

Office: Garden City, N. Y.; Factories: Garden City and Buffalo, N. Y.



MANUFACTURERS' SPECIFICATIONS ON AMERICAN CO.

THE TABLE BELOW IS BELIEVED TO BE ACCURATE:

AIRCRAFTS AND SEAPLANES AS COMPILED BY AVIATION

NOT BE RESPONSIBLE FOR THE FIGURES GIVEN

[illegible][illegible]

TTPs

1000

ENCLOSURE. (See engine valve control)

[illegible][illegible]

1. The first step is to identify the problem.
 2. The second step is to define the problem.
 3. The third step is to analyze the problem.
 4. The fourth step is to develop a solution.
 5. The fifth step is to implement the solution.
 6. The sixth step is to evaluate the solution.
 7. The seventh step is to monitor the solution.
 8. The eighth step is to maintain the solution.
 9. The ninth step is to improve the solution.
 10. The tenth step is to document the solution.

Copyright © 2004 by John Wiley & Sons, Inc.

1. The first step is to identify the problem or question that needs to be addressed. This involves understanding the context and the specific requirements of the task.

Journal of
Health Politics, Policy and Law

1000

[illegible]

STANLEY

PROSTATE

1. ☐ Yes
 2. ☐ No
 3. ☐ Not sure
 4. ☐ Don't know
 5. ☐ Other

6. ☐ Yes
 7. ☐ No
 8. ☐ Not sure
 9. ☐ Don't know
 10. ☐ Other

11. ☐ Yes
 12. ☐ No
 13. ☐ Not sure
 14. ☐ Don't know
 15. ☐ Other

16. ☐ Yes
 17. ☐ No
 18. ☐ Not sure
 19. ☐ Don't know
 20. ☐ Other

21. ☐ Yes
 22. ☐ No
 23. ☐ Not sure
 24. ☐ Don't know
 25. ☐ Other

26. ☐ Yes
 27. ☐ No
 28. ☐ Not sure
 29. ☐ Don't know
 30. ☐ Other

31. ☐ Yes
 32. ☐ No
 33. ☐ Not sure
 34. ☐ Don't know
 35. ☐ Other

36. ☐ Yes
 37. ☐ No
 38. ☐ Not sure
 39. ☐ Don't know
 40. ☐ Other

41. ☐ Yes
 42. ☐ No
 43. ☐ Not sure
 44. ☐ Don't know
 45. ☐ Other

46. ☐ Yes
 47. ☐ No
 48. ☐ Not sure
 49. ☐ Don't know
 50. ☐ Other

51. ☐ Yes
 52. ☐ No
 53. ☐ Not sure
 54. ☐ Don't know
 55. ☐ Other

56. ☐ Yes
 57. ☐ No
 58. ☐ Not sure
 59. ☐ Don't know
 60. ☐ Other

61. ☐ Yes
 62. ☐ No
 63. ☐ Not sure
 64. ☐ Don't know
 65. ☐ Other

66. ☐ Yes
 67. ☐ No
 68. ☐ Not sure
 69. ☐ Don't know
 70. ☐ Other

71. ☐ Yes
 72. ☐ No
 73. ☐ Not sure
 74. ☐ Don't know
 75. ☐ Other

76. ☐ Yes
 77. ☐ No
 78. ☐ Not sure
 79. ☐ Don't know
 80. ☐ Other

81. ☐ Yes
 82. ☐ No
 83. ☐ Not sure
 84. ☐ Don't know
 85. ☐ Other

86. ☐ Yes
 87. ☐ No
 88. ☐ Not sure
 89. ☐ Don't know
 90. ☐ Other

91. ☐ Yes
 92. ☐ No
 93. ☐ Not sure
 94. ☐ Don't know
 95. ☐ Other

96. ☐ Yes
 97. ☐ No
 98. ☐ Not sure
 99. ☐ Don't know
 100. ☐ Other

101. ☐ Yes
 102. ☐ No
 103. ☐ Not sure
 104. ☐ Don't know
 105. ☐ Other

106. ☐ Yes
 107. ☐ No
 108. ☐ Not sure
 109. ☐ Don't know
 110. ☐ Other

111. ☐ Yes
 112. ☐ No
 113. ☐ Not sure
 114. ☐ Don't know
 115. ☐ Other

116. ☐ Yes
 117. ☐ No
 118. ☐ Not sure
 119. ☐ Don't know
 120. ☐ Other

121. ☐ Yes
 122. ☐ No
 123. ☐ Not sure
 124. ☐ Don't know
 125. ☐ Other

126. ☐ Yes
 127. ☐ No
 128. ☐ Not sure
 129. ☐ Don't know
 130. ☐ Other

131. ☐ Yes
 132. ☐ No
 133. ☐ Not sure
 134. ☐ Don't know
 135. ☐ Other

136. ☐ Yes
 137. ☐ No
 138. ☐ Not sure
 139. ☐ Don't know
 140. ☐ Other

141. ☐ Yes
 142. ☐ No
 143. ☐ Not sure
 144. ☐ Don't know
 145. ☐ Other

146. ☐ Yes
 147. ☐ No
 148. ☐ Not sure
 149. ☐ Don't know
 150. ☐ Other

151. ☐ Yes
 152. ☐ No
 153. ☐ Not sure
 154. ☐ Don't know
 155. ☐ Other

156. ☐ Yes
 157. ☐ No
 158. ☐ Not sure
 159. ☐ Don't know
 160. ☐ Other

161. ☐ Yes
 162. ☐ No
 163. ☐ Not sure
 164. ☐ Don't know
 165. ☐ Other

166. ☐ Yes
 167. ☐ No
 168. ☐ Not sure
 169. ☐ Don't know
 170. ☐ Other

171. ☐ Yes
 172. ☐ No
 173. ☐ Not sure
 174. ☐ Don't know
 175. ☐ Other

176. ☐ Yes
 177. ☐ No
 178. ☐ Not sure
 179. ☐ Don't know
 180. ☐ Other

181. ☐ Yes
 182. ☐ No
 183. ☐ Not sure
 184. ☐ Don't know
 185. ☐ Other

186. ☐ Yes
 187. ☐ No
 188. ☐ Not sure
 189. ☐ Don't know
 190. ☐ Other

191. ☐ Yes
 192. ☐ No
 193. ☐ Not sure
 194. ☐ Don't know
 195. ☐ Other

196. ☐ Yes
 197. ☐ No
 198. ☐ Not sure
 199. ☐ Don't know
 200. ☐ Other

201. ☐ Yes
 202. ☐ No
 203. ☐ Not sure
 204. ☐ Don't know
 205. ☐ Other

206. ☐ Yes
 207. ☐ No
 208. ☐ Not sure
 209. ☐ Don't know
 210. ☐ Other

211. ☐ Yes
 212. ☐ No
 213. ☐ Not sure
 214. ☐ Don't know
 215. ☐ Other

216. ☐ Yes
 217. ☐ No
 218. ☐ Not sure
 219. ☐ Don't know
 220. ☐ Other

221. ☐ Yes
 222. ☐ No
 223. ☐ Not sure
 224. ☐ Don't know
 225. ☐ Other

226. ☐ Yes
 227. ☐ No
 228. ☐ Not sure
 229. ☐ Don't know
 230. ☐ Other

231. ☐ Yes
 232. ☐ No
 233. ☐ Not sure
 234. ☐ Don't know
 235. ☐ Other

236. ☐ Yes

100

1

Fuselage Analysis

Continued from page 1455

tail. There are seven conditions for which the fuselage must be analyzed. These are, high incidence, low incidence, maximum elevator and elevator load, maximum in and rudder load, level landing, three-point landing, and nose-up. The above fuselage structure need not be analyzed for all these conditions. Experience and judgment will indicate which conditions will impose the greatest loads on a certain portion of the fuselage. For example, the maximum tail loadings need only to be investigated as far forward as the rear of the lift train. Some additional stress investigations are also needed for members carrying elevator or landing loads in addition to the loads obtained in them in the analysis for the fuselage conditions.

High Incidence Condition.—In this condition, the fuselage is assumed to be subject to the following loads:

a. Weight of fuselage and contents multiplied by the high incidence load factor. These forces are assumed to act normal to the propeller axis.

b. The high incidence wing reactions multiplied by the factor required for equilibrium.

c. A load on the nose of the fuselage, acting along the propeller axis, and of the magnitude required by the conditions of equilibrium. This load is assumed to represent the resultant of the propeller thrust, drag of the fuselage, and the components of the weight and inertia forces of the fuselage and contents parallel to the propeller axis.

d. An arbitrary load at the tail post, acting normal to the propeller axis, of the magnitude and acting up or down as required for equilibrium.

Low Incidence Condition.—The analysis for this condition is almost exactly similar to that for high incidence. The loads are applied in the following manner:

a. Weight of fuselage and contents multiplied by the low incidence load factor. These weights are assumed to act normal to the propeller axis.

b. A load at the tail post, or on the horizontal tail surfaces perpendicular to the propeller axis, and of the magnitude required to obtain equilibrium.

c. Loads from the wings equal to the forces representing the weight in the low incidence condition, multiplied by a factor sufficient to make the sum of the forces perpendicular to the propeller axis equal to zero.

d. A load along the propeller axis sufficient to make the sum of forces parallel to this axis equal to zero. This load is placed at the nose of the fuselage.

The loads and factors under a, b, c, and d of both high and low incidence must be obtained by applying the equations of equilibrium to the entire structure. This will be done later for our purpose in the high incidence condition and the method employed illustrated. For various reasons the wing reactions on the fuselage do not exactly balance the weights and inertia forces of the fuselage. They are therefore increased by a factor K to obtain equilibrium. Before the analysis of the fuselage drawing can begin the entire system must be in static equilibrium—when means summation V, summation H, and summation Moments, all equal to zero. This is especially true of us are to solve the stress graphically also our diagrams will not show. The analysis will be made graphically for high incidence to detail later in the chapter.

Maximum Rudder and Elevator Load.—The fuselage is to be analyzed for a load on the tail of the magnitude specified in Table 17. This load on the tail is assumed to act down. For planes without tail slide such as airplanes the

fuselage must also be investigated for as up load as the tail. The investigation for these conditions need not be carried farther forward than the rear of the lift train.

Maximum Pin and Rudder Load.—The fuselage must be analyzed for maximum loads on the vertical tail surface as specified in Table 17. These loads must be assumed to act either way. In analyzing for this condition, proper allowance must be made for the increase due to the fact that the rate of pressure on the vertical surface is above the fuselage, which is the case in nearly all present day planes. The usual method of taking care of this increase is to apply a load point, at the tail end of the upper trim and a load in the opposite direction on the lower, the resultant of these two loads being equal to the load on the tail. This will be discussed later for our plane. This condition severely determines the top and bottom trim members of the fuselage. Where there is no left open for complete proper provision must be made for carrying through the loads such as greatest the center. Where the top and bottom trims are similar it will usually be found easier and safer to analyze only the top trim for its load and assume that the member members of the bottom trim carry the same load. This is safe because by our method of caring for the forces we have applied a much greater load on the top trim than on the bottom.

Level Landing Condition.—The fuselage is assumed to be subject to the following loads:

a. Weight of the airplane, except the chassis, assumed to act normal to the propeller axis.

b. The landing gear reactions from the level landing chassis analysis. The vertical components of these reactions should balance the weight.

c. Horizontal forces loads representing the inertia force due to slowing down of the airplane. If desired the parallel loads of the fuselage weights may be assumed to act parallel to the resultant of the landing gear reactions. In this case the reactions should balance the sum of the loads. Sometimes moments must also be used, of course, before the system is in equilibrium and can be analyzed. Due to the position of the center of gravity shifting with and without landing gear and slight resistance in placing loads at point points etc. it will usually be found that the reactions will not equal zero even though the reactions equal the sum of the parallel loads. The individual reactions of the chassis struts on the fuselage should be slightly altered in this case until equilibrium is obtained. This condition will not be critical for the rear half of the fuselage.

Three-point Landing Condition.—In this condition the plane is resting on the wheels and tail with all shock absorbers made fully deflected just as in the chassis analysis. The fuselage is assumed to be subject to the following loads:

a. Weight of the airplane assuming only such weights as are carried by the landing gear without the aid of the fuselage. These loads are assumed to act perpendicular to the ground line.

b. The landing gear reactions, and tail slide load, which is determined from the landing gear analysis should balance the weights. The entire fuselage side truss will usually have to be analyzed for this condition. As before complete equilibrium will not be obtained and slight variation of the three-point reaction may be needed. In analyzing the landing gear we should have used the center of gravity location, wheel chassis position, and then use forces would balance correctly.

Wing Over Condition.—The front part of the fuselage must be designed to resist the forces to which it would be subject in wing over when the landing gear is of such construction that the probability of such an accident is remote. To analyze this condition the airplane is assumed to be resting on the wheels and the center of the propeller back so that por-

tion of the primary structure of the fuselage that would exist in the ground line. The gross weight of the airplane is assumed to act at the center of gravity and perpendicular to the ground line. The load factor employed for this condition is the same as that for three-point landing.

Maximum Conditions

In addition to the seven foregoing conditions the fuselage structure must be investigated for several other major conditions. These are:

1.—**Turns Load.**—The torque transmitted to the engine mount may be computed by $T = 60,000P/V$ where T is the torque in inch ft. P is the horsepower of the engine V is engine speed in r.p.m. (For geared engines use gear speed.)

For motop with parallel engine beams or extremely and for motor mounted engines the torque load on each engine beam is T/V where d is the distance in inches between the beams. This load acts down on one engine beam and up on the other. For radial engine mounts the torque load is equal to T/V where V is the radius of the mount ring. The same radial mount ring has four points of attachment to the engine mount base. The torque load T/V is divided equally in four parts and one part applied tangent to the mount ring at each point of attachment. These forces must all be made to point in a clockwise direction. The analysis must be made through to the wing connections.

A load factor of two is assumed for non-drive engines and three for geared engines. The loads found at the motor-beam for these conditions are to be added to the loads found in the high incidence condition and the member designed on the basis of the two loads. The loads are to be added arithmetically and not algebraically, that is to say the sum of the two loads will always have the same sign as the high incidence load but in the first place. The reason for this is that in the high incidence condition the member on the other side of the fuselage from that we are dealing with will usually have the same load but of the opposite sign.

Stress Investigation.—The stress must be investigated for both the high and low incidence conditions assuming that 75 per cent. of the design load is acting on one wing and 25 per cent. on the other. For stress examination this investigation should be carried on for the engine surrounding the cabin.

New Engine Mount for Rotating Airplane.—Very often one plane is designed to be adapted to several engines. The same mount will take all the engines so several mounts must be designed. Rather than require a complete new analysis for each mount the Commerce Department has outlined an alternate procedure. The following conditions must be investigated:

1.—High incidence and torque.—The analysis should be made as specified for these conditions except that the reactions required may be assumed to be applied at the points where the reactions are placed in the structure of the engine mount. 2.—Low incidence.—The load of each member should be assumed to be of the opposite sign to the load in that member, due to high incidence and torque, and its magnitude equal to 75 per cent. of the load. This condition shall be assumed to be acting over as well as inverted flight, the high ratio of weight of load being used so this will be possible.

3.—Low load.—The mount should be investigated for the weight of the contents acting sideways with an ultimate load factor of one-quarter of the high incidence factor.

4.—In design for which the landing load factor is greater than the high incidence factor, the landing gear should be in place of the high incidence factor in the analysis of the high incidence and torque conditions.

It must be borne in mind that the engine replacing each other in any plane must be of nearly the same power. It is usually necessary to reduce the weight and the load factors

MODERN FLIGHT

STUDENT PILOTS and young men about to enter the game see out down the number of hours usually necessary before adding by a study of the greatest flying instruction book of the year, Modern Flight.

CHIEF PILOT CLEVELAND, out of the wealth of his 16 years flying experience explains in simple statements language every maneuver of the controls for takeoff, landing, straight flight and aerobatics.



SAVES YOU \$50.

when you LEARN TO FLY

The author of every flying student is to be able to handle a ship alone in the least time possible. Purchase of Cleveland's Modern Flight is a long step in that direction.

HERE IS MY DOLLAR

Order your copy of Cleveland's complete flying series.

Name _____

Address _____

Manufacturers of the Standard No. 401, Alexander Industries Bldg., Denver, Colo.

The Airmasters



The Elegance and Comfort

of an expensive limousine, with the lightness and easy control of an open-cabin ship are found in

The Airmaster Coupe and the Youngtown Youngster "The Little Airmaster"

Write—or wire—for full information on these three- and two-place personal flying ships.

PRICES (Subject to Change)
\$15,500 and \$23,375
with Leiford Engines.

A Few Desirable Distributors' Territories Open

The Ohio Aero Manufacturing Corporation
Youngstown, Ohio

where a new engine is installed. The new weight from the new load factor may not be greater than the engine from the old factor also a complete new analysis of the plane is necessary. The designer must also be careful that he has moved his center of gravity location appropriately. Moving the center of gravity not only means a new fuselage and other components is necessary but it also will affect the other portions of the plane.

Load Factors and Tail Surface Loads

In the analysis of the fuselage for the flight conditions to factors used in the analysis of the wings for these conditions must be employed. For loading conditions the factor used in the design study must be used. For analysis only the loading factor is used. For the maximum tail load condition the required load may be obtained from Table 17.

Table 17.

Class	Load per sq. ft. Airframe and Horizontal Tail Surface	Load per sq. ft. Vertical Tail Surface
2000 lb. or less	30	22.5
3000 lb.	20	35
10,500 lb.	15	31.25
Over 15,500 lb.	15	31.25

For our plans the factors and loadings are as follows:
High Landing — 4.0 Low Landing — 4.0
Level Landing — 4.0 Three-point Landing — 4.0
Maximum Steady and Climax Load — 30 lb. per sq. ft.
Maximum Pin and Rudder Load — 32.5 lb. per sq. ft.
Wing Over — 4.0

Analysis of Conventional Monoplane Fuselages

Fig. 71 is a line diagram of the fuselage of the Conventional Monoplane that we are analyzing. The top drawing is a plan view in which we are looking down from above, the middle drawing is the side view as we ordinarily see the plane and the bottom view is looking up from under the fuselage. The lines of the diagram of Fig. 71 represent the outlines of the tubes that form the fuselage structure. It is to be noted that except in one case the entire structure is built up of triangles of tubes. A triangular arrangement is built up and determined—that is the basic in such manner as to build strength. In the one location where there is a triangle in directly in front of the pilot where a cross strut would affect his vision—the four corners have been joined to make the frame rigid.

The diagonals indicated on the plan view by dot and dash lines go from the lower wingpost on one side the fuselage to the upper wingpost on the other side and then back to the original wingpost but one bay further on. These diagonals are necessary to make the rear half of the fuselage rigid against torsional loads. They stay at the rear of the tube as it is desirable to keep the entire free of obstruction. Every part of the upper and lower corners of the fuselage carrying the other must be well-gauged to keep the structure rigid. This is easily done on small planes such as ours but on large planes it is necessary to use part of the cabin space for strengthening. The one frame which it has not been possible to indicate in Fig. 71 is that directly behind the engine gear and made up of members 15 and 16. This frame acts as a heavily braced or else less diagonals running across it. In this particular location the twisting of the engine is not severe and would distort an advanced frame.

It will be noted that in the plan and bottom views the top portions of the fuselage are kept straight. Directly the portions of the fuselage have comparatively small loads and the section near the cabin is reached. The load at this section gives a fairly small tube for a fuselage and this tube is usually carried right through to the tail post even though it is very strong strength. If the fuselage is very strong, then a great many welded joints are saved and a single jig is all that is necessary. The fastening out of the

top in the last bay in the side view is to give the struts a horizontal support.

The top and bottom members are both horizontal in the neighborhood of the cabin. At this region the wings and chassis are attached to the fuselage and if the fuselage are horizontal in at these views as we have them the design of the corner fittings is greatly simplified. It also gives a square area which for a cargo or baggage carrying plane is a great advantage, as all loads and stresses can be utilized and the full volume of the plane made use of.

The short dashed lines indicate the frames of the pilot's seat. By this arrangement a one-piece windshield can be used. No quartering or cross-bracing of the frame is necessary in these lines are not structural members and carry no load but the weight of the window.

The type of truss used is known as the Warren truss. For the type of plane it is the lightest and involves the fewest members. The members are numbered for future reference, namely, when the various loads under each condition for each member are listed in the chart at the completion of the design analysis.

In the next chapter the fuselage weights will be distributed along panel points and the analysis of the fuselage completed for several of the conditions.

Copyright Alexander Klimin

To be continued in the next issue of AVIATION

Etching of Aluminum Alloy Propellers

Continued from page 1477

ended in 10 to 15 sec. This is not recommended, however, since small covers on some cause a large variation in the depth of etch and there is danger of over-etching.

The importance of controlling the temperature of the etching solution is shown in Fig. 1. The depth of etch was measured by the loss in weight of standard samples from aluminum alloy propellers. These samples were subjected to the action of 5, 10, and 20 per cent. sodium acetate for a period of 5 min. The temperature was varied from 50 to 150 deg. F. Within this temperature range an increase of 1 deg. F. is equivalent to increasing the concentration of the etchant solution 1 per cent. When the temperature is below 50 deg. F. the time required for the proper depth of etch increases very rapidly and it has been found that long etches in acidic or cold solutions of organic acids produce an unsatisfactory etched structure. When the temperature is greater than 150 deg. F. there is danger of over-etching as noted above.

The strength of the etching solution decreases with use, and to preserve effectiveness should be made for this in regulating the loss of acid and the temperature. It has been found that the 10 to 20 per cent. solution recommended gives very good results provided the other variables (time and temperature) are controlled. Below 50 deg. F. changes in the concentration in the above range have practically no effect upon either the depth of etch or etched structure. Between 50 and 100 deg. F. the rate of attack is about twice as great for the 10 per cent. solution as it is for the 20 per cent. solution. For 70 deg. F. the effect of increasing the concentration from 5 to 20 per cent. at various temperatures is the same, 60 to 150 deg. F. The loss in weight shown in the graph is a measure of the depth of etch and was determined by weighing standard samples before and after immersion for five minutes in the solutions indicated.

The Airsedan



For The Discriminating Purchaser:

THE AIRSEDAN

AIRLINE OPERATORS will find this plane meets with all their requirements. The cabin has exceptionally comfortable seats for four large passengers and the pilot's visibility is unexcelled.

PRIVATE OWNERS will appreciate of the fine appointments which are selected to satisfy the most critical taste.

CORPORATIONS desiring to keep step with the progress of the times will find that here is an efficient and up-to-date vehicle for transporting executives and personnel to widely separated branches.

We will be glad to assist you in determining the adaptability of this fine product to your requirements.

BUHL AIRCRAFT COMPANY
Marysville Michigan

Immediately upon removing the propellers from the caustic solution they are plunged into water, which should be at approximately atmospheric temperature. During this operation, care should be observed not to allow the caustic to dry as it affects the final appearance of the etched surface. After rinsing, the propellers are transferred to the acid bath. The



Fig. 5 (left)—Photograph of hub section of detachable-blade aluminum alloy propeller, light-etched. Arrows point to cracks formed during forging. Fig. 6.—Photograph of a section of a detachable-blade propeller, light-etched. Arrows point to cracks formed during whirling test.

removal of the copper deposit may be hastened by agitation or by swabbing with a fiber brush. After all traces of the copper deposit are removed the propellers are transferred back to the same rinsing water used to remove the caustic solution. It is desirable to use the same rinsing water for both the caustic and acid dips for the reason that the same water may be used for long periods of time due to the acid

and caustic controlling each other; otherwise, it is a necessity to change the water quite frequently. Final washing is preferably done in hot water, and for convenience it is one water used for preheating the propellers may be used. A thorough washing they are allowed to dry in air or in a vacuum oven. The final appearance of the hubs should be silvery white and free from all stain or smut.

An alternate method of etching which is applicable when only a few propellers are to be etched is as follows: After the propellers have been prepared for etching as described above, a few gallons of caustic solution are made up and applied to the surfaces rapidly and evenly with a wash or brush. For this purpose the concentration and temperature should be near the lower limits recommended for the same. Each of the surfaces of the caustic is then washed in an attempt to obtain an even etch over all portions of the blade due to the time that is required to apply the caustic to the different portions. After the caustic has remained in contact with the propeller a sufficient length of time to produce the desired depth of etch, it is removed by swabbing with cold water or washing in running water. The etch is then rinsed freely with a wash. It is important to have the final water and acid solution quite cold to prevent streaking and staining. The surfaces are then thoroughly washed with alcohol, hot or cold water, and dried.

For the inspection of the etched surface, a magnifying glass with a power of two to four diameters should be used. A reading glass may be used in the absence of a better glass. The sections adjacent to the hub and especially the bottom of the fillets and areas adjacent to both hubs should be examined very carefully and the slightest indication of an imperfection or flaw etched will be noted. The areas thus marked will be rubbed with a fine grade of sandpaper or emery cloth until 00005 to 00001 in. of the surface metal has been removed. This operation should be followed by the

application of a few drops of a weak concentrated aqueous solution of caustic soda. (The solution may be prepared by filling a wide-necked bottle one-half full of hot water and adding caustic until an excess is obtained.) The solution may be applied to the crack or apparent defect with a small wash, put a compound, or a splinter of wood. After the caustic splinter has remained in contact with the metal for one or two minutes the excess caustic is wiped away with a cloth saturated with water or a weak acidic and solution. If the crack or defect has continued beneath the surface it will show as a definite black line, and upon successive use of the technique may be seen to form in the defect if it is a crack or seam. If necessary, a new application of caustic may be made and an etch of any desired depth can be made by further adding the caustic a few drops at a time and removing the copper deposit as described above after each application so as to follow slowly its progress. In the same manner seams or bad marks should be investigated. One example of a crack that led to the failure of a detachable-blade propeller is shown in Fig. 3. This crack would have easily escaped detection due to its position at the bottom of the fillet between the marked section of the hub and the blade. Scratches, bad marks, or other indications of any ap-

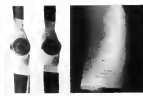


Fig. 3 (left)—Photograph of Curtiss-Road "D" type propeller showing typical cracks formed during service of the propeller. Fig. 5.—Photograph of Curtiss-Road "D" type propeller, failure due to notch effect of filletting.

pearance should be carefully worked out with an emery stone, whenever they are found in any section of the propeller.

Aluminum alloy forgings used in the manufacture of propellers are manufactured in a general orientation covering all aluminum alloy forgings. This specification requires that the forgings be free from blisters, fins, seams, laps, cracks, scrapings, and other defects that affect their value. Defects that escape detection in the forging blank may be revealed by the etching of the finished or machined propeller, and in addition the etched propeller may show discoloration caused by oil or oxide being forced into the metal by the hammer and marks of considerable depth scratching the surface not accompanied by actual discontinuity of the metal. When a section of a wrought or forged aluminum propeller shows cracks formed during the forging operation or blisters as the result of repeated strains to which the metal is subjected in service, there is little doubt as to the effect on the strength of the propeller. The same is true of scales and surface markings of non-metallic substances, scrapings, seams, scrapes, and blowholes. It must be borne in mind, however, that all forged aluminum alloys contain these latter defects to a slight degree. In interpreting the results revealed by

Aviation Fundamentals made easy

THOUSANDS of young men and women have taken their first step in aviation by studying *Our The ARPA Home Study in Aviation*.

The ARPA Home Study in Aviation is a 32 page book, with blue print of model airplanes, and a list of 25 questions to be answered by the student.

Endorsed by leading aviators and students. Professor Alexander Klumpp, Drexel Guggenheim School of Aeronautics, says:

"I have not had an opportunity of looking through *The ARPA Home Study Course in Aviation*. It is clear, sound and well prepared, and an excellent introduction to practical aviation."

Col. Harold B. Hartney, American "Ace" says: "I must congratulate you on your Home Study in Aviation. It gives an excellent basis for anyone contemplating either the business of aviation or military service in the field."

Floyd J. Logan, Aviation Journalist, Cleveland, writes: "Your Course gives to the beginner, in simple language, answers to questions which arise in the minds of beginners."

Lo. Leigh Wade, Round-the-World-Flter, says: "Your Course is a wonderful exposition of the fundamentals of aviation. I can readily recommend it to those seeking solid and interesting information."

The price of the course complete is

\$1.00

Send check, money order, or write for information, special price to manufacturers, clubs and schools.

A Non-Profit Organization

American Institute for the Promotion of Aviation, Inc.	
1015 Fifth Avenue, New York City (City Office, N.Y.)	
Domestic:	
Enclosed find one dollar. Please send me the ARPA Home Study in Aviation.	
Name	_____
Street	_____
City	_____
State	_____
Zip	_____

Madison Lights Up

Penrose Field, at Madison, Wisconsin, is on the Chicago-Twin City Airway. Its wide runways form a huge letter 'T', 2800 by 1200 feet. Although Penrose Field is typical of many American airports, it has special significance because it is well lighted for night service.

The lighting equipment includes seventeen boundary lights, seven 120-degree floodlights, a rotating beacon, wind cone and hazard lights, and control apparatus—all furnished by the General Electric Company, a pioneer in the manufacture of airport lighting equipment.

Complete Lighting Equipment for Airports and Airways

Aluminum Floodlights

Beacons

Control Apparatus

Wires and Cables

GENERAL ELECTRIC

GENERAL ELECTRIC COMPANY, SCHENECTADY, N. Y. SALES OFFICES IN PRINCIPAL CITIES

sliding, expansion and adjustment caused by prolonged operation on various types of structures are necessary to their growth between defects that are of a nature that will lead to failure of the material and those which are more or less characteristic of all aluminum alloys.

The examination of a large number of propellers that have failed both in service and when on test has shown that the failure has always been the result of a crack gradually spreading. There have been few instances of failures that can be definitely attributed to faulty material, and the method of inspection outlined herein will enable relatively inexperienced personnel to detect cracks before they spread to proportions that will cause failure under ordinary operating conditions.

The photograph shown in Fig. 4 is an example of a lap formed during forging. In this particular instance the defect is not serious, but in some instances or around the top extended clear across the blade and was the cause of the propeller failing on the test stand. Blows are usually found on the reverse surface of the blade on the both surface and usually extend parallel to the longitudinal axis of the propeller. They are simply limitations uncovered during the machining operation and are more or less characteristic of all aluminum alloy forging alloys. Where the scars are numerous and of considerable length the material may be considered inferior. It seems they may be easily distinguished from a crack by the nature of the edges which are even and often covered with a film of oxide.

Cracks are of two classes: those formed during the forging operation, an example of which is shown in Fig. 5, and those formed during the service of the propeller, which are shown in Fig. 3, 6, and 7. It is the latter type of crack that is of most concern to the designer. They are pronounced enough to be easily detected and cannot be mistaken for scars or other types of defects. The former type was found

in one lot of propellers that were manufactured on an improperly designed die, but were detected before they had been put into service. They had no effect on the service of the propellers and required no appreciable repairs. Cracks have also started in rotation, such as produced by vibration, and at the base of sharp fillets. Examples of these are given in Fig. 8 and 9, respectively.

In examining reports of propellers that failed on the test stand, it is interesting to note that in practically all instances the type of crack shown in Fig. 6 is nearly always associated with bolts or excessive vibration.

A light nick, if properly made, has very little effect on the balance of the detachable-blade propeller as determined against a standard blade. If the tuning is properly checked and the temperature and concentration are maintained, no change of balance occurs in propellers that are etched slightly in service.

The removal of the anodized or hardened film of oxide from the leading edge causes them to be less resistant to erosion by sand or rain. This can be overcome by applying a varnish along the edge of the propeller or by covering with enamel. Where night flying is practiced it is desirable to paint the front surface of the blade with varnish until the film is in place on the tip. For ordinary flying there has been no serious objection to glare on the part of the flying pilot.

It is advisable to protect the surface of the propeller when in either storage or service by covering with a light oil and frequent polishing of the propeller, in connection with the maintenance of the airplane, with a good commercial metal polish such as used for silver or brass, is most desirable since it furnishes an additional means of inspection for cracks. The propeller should be examined and subjected to a light oil and thorough inspection after each movement of 100 to 200 hr. of flying.

Edo Pontoons and the Ryan Brougham

THE Ryan Brougham, the ship made famous by Lindbergh and made popular by its own carrier by its own carrier by its own carrier, proved its quality in a Seaplane. And what a Seaplane! With an exceptionally smooth interior and superb the most perfect performance of the Brougham hardly affected by the weight of floats, it becomes a natural leader in the class of Wright engine Seaplanes.

With its wonderful performance the Ryan Brougham Seaplane combines the most practically essential for commercial work. Two side doors of the floatage, in conjunction with its wide double doors, provide a most convenient means for passengers getting in or out. Seaside drift at the water and on the low allows the ship to be brought close



to a beach, upon making embarking and disembarking more comfortable.

Edo straight steel floats with trailing wing tips, and that is an important feature in Seaplane operation. Good seaplanes are judged by the placement of the floats, giving better buoyancy of 120%. High buoyancy of the floats, greater wing area, the straightest floats, or any needlessly high deck, poor or large like a motor boat, built of "Alcoa" and Aerobically tested

the floats must conform and cover it or each float must allow the Seaplane to be moved in open and eliminating of the trouble of frequent launching which would hinder operation.

EDO

EDO AIRCRAFT CORPORATION
COLLEGE POINT, L. I., N. Y.

STANDARDIZED ALL METAL SEAPLANE FLOATS

The Stearman Biplane

Continued from page 1474

of the upper wing curves an aileron operated by torque tubes (one for center section, then by push-pull tubes to the control gear in the cockpit). By suitable strut calibrations, a differential action has been secured, thus materially reducing its yaw due to roll. A single tank of 30 gal. capacity, with a pressure gauge, occupies the center section.

Fuselage of Pratt Type of Truss

The fuselage is of the Pratt type of truss with main members. It is produced from chrome molybdenum and made in section steel tubing. It will be recalled that elsewhere in the article the statement was made that the structure all the way with the same rigidity of the power plant. To secure this arrangement a detachable engine section has been provided and hence all members are welded on separate gusset plates carry an exact duplication of the Struss used at the forward station. All fittings and chassis fittings are pig welded into later introduced into the main assembly, which is a truly completely pig welded with an allowable tolerance of the tail part of 1/16 in. Within the bay normally occupied by the passenger the upper legpiece, on the left side, has been replaced by cross tubes, thus facilitating entrance and exit to that compartment. The controls which are permitted in the rear cockpit are duplicated in the forward cockpit but completely removable. To enable the pilot to be comfortable under varying conditions and to have the proper motion, a parachute type seat has been made adjustable through a range of 4½ in. The most important equipment available and is a motor-driven air speed indicator and engine are included. Fuel valves and the gasoline strainer are located in the pilot's cockpit. Much discussion has been raised around the question of proper location of the thrust



Potential and Future Prospects

Yes, All Metal but the Tires

Model "C" OX3, 42950

Write for details

Lenert Aircraft Co.

Pontiac, Michigan

Three Thousand Aerial Views! Fifteen Hundred Live Accounts! Business established in 1922! and Goodwill For Sale!

During the past six years this company has built up a profitable business in aerial photographic work in New England (Mass., Maine, Rhode Island, Conn. and New York). This is an exceptional opportunity to buy a going business in Aviation.

Write Box 885-Q, Aviation



The tail before and after assembly of the Stearman biplane.

It used to appear both parties this manufacturer has provided as permanent equipment a double quadrant on each side of the pilot's cockpit.

Roll being so manipulated in various ways and welded a steel in the engine. The stabilizer, constructed of two bars, is also adjustable at the leading edge by means of a wedge secured from a quadrant in the rear cockpit. By moving in adjustment for due to the effect of torque and wing dip stress can be controlled and with the balanced and satisfactory directional control and stability with free motion is secured. The welded construction is such that the stress for static and dynamic longitudinal stability appear a desirable characteristic.

Safer!

—this new
tie rod terminal



Tighten the lock nut as much as you will—you cannot hurt the terminal or break the tie rod. New, stronger, safer, better. Look out screws on to the terminal instead of on to the rod, avoiding torsional and tensile strains, giving closest thread fit. Write for detailed information. MacWhyte Company, 2905 Fourteenth Avenue, Kenosha, Wisconsin.

MACWHYTE
Streamline and Round Tie Rods
Patent Safe Lock Terminals

Protect your aviation investment —

A DEQUATE insurance at reasonable rates is one of the vital factors in profitable commercial flying.

WE have built up an aviation insurance business through knowledge based upon flying experience and service to manufacturers, operators and pilots.

Our Policies are Written to Cover
Your Individual Needs

Fire and Transportation, Public Liability, Property Damage, Personal Accident, Business Interruption, Commercial Aircraft, Loss of Aircraft, Pilot's Compensation, Medical Expenses, and Uninsured's Compensation.

Let us know your Problems.
Our Consultants Give us Free.

JAMES E. MALLETT & SON
TORRINGTON, CONNECTICUT

at the conclusion of the run. No mechanical adjust into replacements were necessary.

Production of these engines is now under way in the factory which is an completely equipped that army can a, engine accessories and design designs are produced within the plant. The factory is equipped to cast iron, aluminum, or bronze. The machine shop contains automatic gear cutting, planing, boring, machine and multiple drill press and other equipment. A complete assembly department has been organized and includes a tool steel and dynamometer.

Company Has Built Marine Engines

The Hellett Manufacturing Co. has built marine engines involving a production of 480 units per year on the basis of machine production such as will be used in producing the airplane engine. There are already 14 agents in the U. S. five in Alaska, two in Canada and others in Hawaii, Tiki, Brazil, Cuba, England, Holland, Sweden, Norway, India, Japan, the Philippines, Australia, New Zealand, Italy and South Africa who are distributing Hellett Marine engines and will be prepared to handle supplies and service on Hellett Aviation engines.

Alfred Hellett, designer and manufacturer of the Hellett engine, built the first airplane engine in 1925. Now that he has returned to this field, he is determined to build the best commercial engine on the market to sell at commercial prices.

The specifications on the Hellett engine, as supplied to the manufacturer, are as follows:

Displacement	5.00
Number of cylinders	4
Deposition of cylinders	45°
Cooling	Water
Power, rated	130 hp
Weight, standard	\$500
Weight, dry without tank or starter	425 lb
Weight per rated hp	3.28 lb/hp
Bore	4 1/2 in
Stroke	5 1/2 in
Displacement	5.00 cu in
Compression ratio	15 to 1
Shafting of mainshaft	1 1/2 in
Overall diameter	48 in
Length overall	37 in

The Curtiss "Robin"

Continued from page 2432

ing and wooden drag struts. The lift strut is arranged to prevent the warping of the wings from the aileron lock which is the cause of the poor lateral control noted in most types of monoplanes.

All of the control surfaces, including the fixed surface



The skeleton fuselage of the new Curtiss "Robin" (224)

are made of small steel tubes welded in torque tubes and are made of thoroughly heat treated.

The fuselage is entirely of welded, chrome-nickel-iron and

panels and, the Warren truss arrangement being used, also, in the case of lower wings.

The ribs due the aileron motion on the lift struts are made in the same way as are the main wing ribs, pressed from a single sheet of Alclad.

Seats for the passengers are side-by-side and can be adjusted in a fore-and-aft direction readily. The aileron is arranged in a staggered arrangement, giving additional shoulder clearance and comfort. For increasing altitude a fixed control can be fitted in the center, not part of the standard equipment.



Illustration of wing of the Curtiss "Robin." Note the aileron control system.

and, when that is used the two seats are fixed up and down at the aft control ribs in the center of the fuselage.

A 20 gal welded aluminum gas tank is set in each wing and the engine may be fed by gravity from either tank. The pilot has the main stick and pedal control, and has a quick folding stabilizer adjustment at his left side. On the instrument board are mounted a tachometer, barometer, oil gauge, altimeter and air speed indicator.

The interior is completely upholstered and pleasantly finished.

The wind tunnel tests on the Robin showed single seated in all attitudes of the plane, and a natural stability which was completely checked by the full flight aspects. The large dihedral of the lift struts and the small dihedral of the main wings added considerably to the natural equilibrium stability at a light wing load.

In view of the fact that the entire section of the fuselage is made up of welded, built for the comfort and the



Model of latest model of the new Curtiss "Robin" when completed.

body of the plane, and for simplicity of construction, the plane showed the very surprising maximum lift drag ratio.

According to Major Robertson the price of the plane will be about \$8,000 complete. This will be for a plane ready to fly and complete in all details. Provision has been made in the structure for "extras" that the prospective owner might wish such as dual control, wheel brakes, lights or additional instruments.

The Robin promises to be a very valuable and popular type for both the private owner and commercial operator.

The American Eagle Knows no Defeat in Performance contests—



Neither has it ever had a Structural Failure

Brilliant performance, amazing maneuverability, rugged, dependable construction together with its striking beauty of design and finish have firmly established the American Eagle as

"Master of the Skies"

With welded monocoque tube fuselage and red surfaces, selected spruce and balsa wood inlaying with exceptional finish, its safety features far beyond Department of Commerce requirements, which is responsible for the remarkable record of never having had a structural failure.

Manufactured under Approved Type Certificate No. 17 issued by the Federal Aviation Commission



2506 East 12th Street
Kansas City, Missouri

THE NAVY DEPARTMENT

hasly invites and will receive until 12 o'clock noon, Thursday, August 9th, 1928:

(One) Competitive designs for a rigid airship of approximately 6,000,000 cubic feet volume. Design to be submitted in accordance with section 90, A to H inclusive, of the Aircraft Procurement Act approved July 2, 1926 (44 stat. part 2, page 780). Price for designs and bid for construction of one and of two rigid airships to be submitted in accordance therewith as provided in subparagraphs A, H, and E, of Section 30 of said Act;

(Two) Bids for one and for two experimental rigid airships of approximately 6,000,000 cubic feet volume, parts and accessories; and

(Three) Proposals for construction of one and of two rigid airships in accordance with Navy Department's specifications.

Circulars giving full detailed information may be had on application to the Navy Department, Bureau of Aeronautics. Curtis D. Wilcox, Secretary of the Navy, April 9, 1928.

AIRPORTS AND AIRWAYS

St. Louis, Mo.

By M. E. Alexander

Air express service between St. Louis and other on most of the western and routes was inaugurated on May 1 under the provisions of a contract between the Eastern Air-
craft Corp., operator of the mail line, and the American Railway Express Co.

For the service all of the mail pilots—Hed Gurney, Lee-
be Smith, and Hays H. Stenger—were at the end of the
route, so that there were surplus planes for the first ship-
ment. A Whitworth Travel Air, a Ryan Dougherty, and a
DHJ were used for the aerobically trip. Express service from
the Chicago and was started on the next day.

The new service gives express patrons aerial transportation
with all the points as the transcontinental air routes as well
as with Dallas and the route on the Boston-New York, Los
Angeles-Lake City, Chicago-Chicago, and Los Ange-
les-Portland routes. The service is being expanded to include
other on every air route in the country.

Special delivery will be given air express mail. The ser-
vice will be added to the telephone to serve as his shipment
has been scheduled and if he is ready for delivery the pack-
age will be sent to him by special messenger. The route
spreads from 35 miles the quarter point, between St. Louis
and Chicago, to 60 miles from St. Louis to Los Angeles.

Ashcraft-Brook, Inc., has joined the ranks of the big busi-
ness organizations that use the airplane. A First Freighter
purchased for \$12,500 weekly by Ashcraft-Brook from the
Van Hoffmann Aircraft Co. Inc., is being piloted by Lam-
bert-St. Louis Field along special routes out of the com-
pany—silver wings and red linings, with the home colors
on the sides—and will be used in several other lines.

Joseph H. Hines, who has been flying out of Lambert Field
for several years, will pilot the plane. According to an
official of Ashcraft-Brook, it is the first of a fleet to be ac-
quired by the organization for the transportation of its
officials and sales representatives. One of the first trips of

the new plane will be to Canada and an "NC" design,
it is to be obtained for the plane from the Department of Air
Service.

Ed Chaslett, president of the General Truck Service Co.,
another business man who uses an airplane in his work, is
an OX-5 Travel Air all ground for a trip to Fair-
fax, Va., where he will confer with officials of the Railroad Ex-
press Co. Fred Gurney, secretary and general manager
and Carl Van Aachen, sales manager of the Richmond ex-
press recently visited the field and inspected Chaslett's plane.
Chaslett himself is a pilot and his wife is taking flying in-
structions.

Cl C. Strader of Long Beach, Calif., flying solo on the
the waters of the Sound Lake peninsula, recently flew
Pacific Airport in a Hino Travel Air. He was accompanied
by Zhang Young, a professional jumper, who demonstrates
the Hino plane to prospective purchasers.

Boston, Mass.

By Daniel Randolph

Boston flying during the first week of May 1938 showed
total of 372 by divided as follows: Naval Reserve Air to
100 on September 27 by 20 men, Boston Airport Corp. 44 by
30 men, Army Engineers and reserves 43 by 30 men, Army
Corp. 34 by 27 men, 321st Observation Squadron, M.S.G. 1
by 50 men, Boy State Flying Service, Inc., 29 by 30 men,
Edward T. O'Toole 17 by 10, Old Colony Airways, Inc., 12 by
Thomas Conn 11 by 10, Colonial Air Transport 10 by 10
men, and individuals 17 by 10.

An Egyptian, Ali Bey Yehia of Alexandria, is taking 10
flying lessons at the Boston Airport from Boy State Flying
Service. This company flies its 20 by 30 men, using ex-
press plane, an American Eagle. Ali has been in the city.

The new model during the first week for the new express
organization came being piloted by Boston Airport Corp. at
its new hotel. Hines, who has been flying out of Lambert Field
for several years, will pilot the plane. According to an
official of Ashcraft-Brook, it is the first of a fleet to be ac-
quired by the organization for the transportation of its
officials and sales representatives. One of the first trips of

The Walter Duesch 1936 colonial reliability test was
Travel Air before has taken a new lease of life. It has been
rebuilt and finished and given a new lease of life, and is now
in back at work for the R.A.C. at the airport.

Reserve engine training at the Squamish air station is
being allowed to start their planes, this being permitted by
the first year now due to guarantee of sufficient plane in
training purpose.

First Inspection Held

The first class of aerial reserve students includes 20 men
from Washington, D. C., 33 from Philadelphia, and one from
Newport, R. I. They are at Squamish for 90 days. On ac-
cessfully completing the primary flight work here they will
go to Pensacola for advanced training and maneuvers. The
first inspection of the reserve Navy students at the War-
renton was recently held at Squamish with nearly 50 pil-
ots present. Capt. Reginald D. Thomas, U.S.N.R., is chief
instructor.

The Massachusetts National Guard Squadron 68 (d) is
shooting from Boston airplanes in 1938 when May. Capt.
H. W. Foster filed a meeting with the rear section of an Old
Hawkeye with Capt. Clarence E. Jones as guest, few days later
Hawkeye for practice in preparation for the Lancaster 100
contest next month.

Colonial Air Transport, Inc., completed all and kept its
first week in May except one, being turned back by bad weather.

May 21, 1938

on the night of May 8 at Hingham, Mass. Incoming mail
to Boston for the week included 233 lb. 5 oz., outgoing 467
lb. 4 oz.

Edward T. O'Toole, the writer, and Mrs. Daniel Hochstadt,
now with Albert L. Edson piloting in the new Deanean Air-
port Farnfield make exception to Martin's Vineyard Ser-
vice. May 8, to inspect the new Martin's Vineyard Air-
port is scheduled this summer by O'Toole and Deputy Sheriff
Tom Wilson of Boston and Edgemoor. The route O'Toole
and O'Toole's Airport at South Duxbury was viewed. It is nearly
fenced with only six acres or so of 22 left and the rolling and
rolling. It will be one of the largest airports in the East.

The Colonel expects to carry on several aerobically work
spending in redwood forest and control of aircraft.
May 17, Boston radio station, is broadcasting aviation talks
every Monday, giving the time free. Speakers on the
line include: R. W. Markes of Colonial, Capt. Frank C.
Dawley, reserve pilot on duty with the First Postal Group,
L. P. Jones of Army of Old Colony Airways, and Edward T.
O'Toole. Every flying group is to be given an evening on
the line last year.

Springfield, Mass.

By Chester Thomas Cole

A two-engine model airplane manufactured by Matthew
March flew more than six minutes in a contest arranged
by the local newspaper recently. It was not what the
American declared to be a record. At any rate, the little ma-
chine, on flying until it was lost from sight in some bushes
and being automatically killed. The model building and
designing proving very popular here this spring and the series
of demonstrations held at the various flying clubs here at-
tracted large numbers of spectators and observers.

Joseph E. Kerrigan, treasurer of Massachusetts Airways of
this city, has invited his fellow delegates to the Democratic
National Convention at Houston, Tex., in June to make the
trip in a Ryan cabin plane which the company has been built.
The various delegates also the Kerrigan plane. It is re-
ported that William S. Bradford, Robert W. King, and John
D. O'Connor, the other delegates, have accepted the invitation.

Direct air mail connection for Springfield will be re-
established, it has been announced, with the signing of an agreement
between the Springfield Airport Corp. and the American
Airlines. The airline, which is now flying out of Springfield, will
operate the development of the York Park Airport site and
at the same time increase its service so that it would be used
as a shipping place for the proposed extension of the Cleve-
land-Springfield line out to Boston.

Front Deane Hall to Hangar

The Springfield Airport Corp. has done some work on the
hangar and it has been used by light commercial planes. The
old deane hall was converted into a large hangar and the
old buildings which were part of the pack's equipment as a
new truck and amusement center are said to be nearly
ready for flying purposes.

The American Airways Corp. under the agreement will
complete the work of grading the field and will construct light-
ing for night flying. The first is 23 ft. from the post office.
Repairs have been laid out to give lengths of 3100, 2500,
and 2200 ft. in various directions.

The company announced at the same time that it out-
fitted a flying service between this city and Philadel-
phia, Pa., from here. A train has been based on that city
to serve as a terminal and plane include also passenger, mail,
airmail, and express service for the Philadelphia area.

New York interests will confer with local interests in the
manufacture here of a new plane soon, it was announced by
the Springfield Airport officials. A factory has been secured



Austin Designs and Builds Aviation Buildings

AUSTIN designs, constructs and equips
A efficient hangars, factory buildings and
other structures at low cost.

Austin guarantees correct construction, quality
and quality in materials and workmanship.
Austin will deliver material and make
material materials for erection by a local building
at a building saving in time, money and expense.
Representative companies from across the
country have used Austin's building service.

Ask for specifications and plans.
"Austin Builds for America."

THE AUSTIN COMPANY

Architects, Engineers and Builders for
the Aviation Industry

New York, Chicago, Philadelphia, Los Angeles, San Francisco, San
Antonio, Dallas, Houston, Fort Worth, St. Louis, St. Paul, Minneapolis,
The Austin Company of Chicago, Inc. and The Austin Company of
the Austin Company of Chicago, Inc. and The Austin Company of Chicago, Inc.

Write for our new catalog No. AV 2

Helmets and
Smart Wearing Apparel
for use in open ship, summer and winter

Special clothing for
CARNIVAL AIR GROUND WEAR

HELMET BROOKS EVERARD CO.

Kalamazoo Uniform Co.
KALAMAZOO, MICH.

OX5 Cylinders

Have your worn cylinders reground and
recoat prices furnished.

Write for prices.
Engine and Machine Part Manufacturing
a Specialty

Edward A. Whaley & Co.
Norfolk, Va.



Chicago crowd rush to greet the Brown crew as the Jan-
kers F-11 which have them from Cleveland come to a stop
on the Municipal Airport. Captain Kitch, Major Fournier,
and Brown crew members are seen leaving the cockpit of
the monoplane.

Keeping Pace... with an Industry that Flies

KEEPING pace with an industry that flies is impossible without a means of obtaining reliable and timely news of events in the industry—concrete authoritative information on technical developments, new planes, engines and equipment—facts about modern production methods, merchandising procedure, operating practice, etc., as well as accounts of other aeronautical developments of importance. A good aeronautical publication should be the means for obtaining the above.

checked and aided by a special news editor. AVIATION news is reliable and complete. It gives the aviation story in national news events. A whole column is devoted to the latest news of the leading airlines.

Technical Articles

IMPORTANT technical articles are included by prominent authorities on the subject. By a technical editor, who is a graduate of a leading aeronautical engineering school. Material is gathered from many sources, both foreign and domestic, and covers the latest developments in aeronautics—aircraft, engines, airframes and engine design, navigation, meteorology, flight flying, etc.

AVIATION is a most favorable medium to present valuable technical articles because of the immense interest it has had with the entire field of aeronautics during twelve years of continuous publication.

New Planes, Engines and Equipment

DETAILED and reliable information on new planes, engines, aerometers, and equipment is published in AVIATION weekly. It is possible to describe practically every aircraft in the world, because AVIATION carries during the course of a month as much aeronautical publication as could be monthly aeronautical publications combined.

Statistics and Specifications

AVIATION was the pioneer in the collection and publishing of statistics on the aeronautical industry. It leads in this field in the present time and publishes much valuable statistical data, including figures on aircraft and engine production, capacity, etc. Most of this is original material collected by the organization.

A table of specifications on American commercial airplanes and airplanes is published in the first issue of each month. It presents, in easily comprehensible form, dimensions, performance data, and other important information, concerning any. A special table for American aircraft engines is published once each month.

Production Methods, Merchandising and Operation

AVIATION specializes in articles which will help those professionally engaged in aeronautics in aviation more effectively, markets more successfully, or operate more economically, as the case may be. It is the business sense of the aeronautical industry. Articles are published regularly on industrial production methods, and progress in the merchandising of aircraft, airlines, aerometers and the selling of commercial aviation services, and on the various phases

of operation, such as altitudes, coast routes, aviation schools and airports.

Departments

THE EDITORIAL pages of AVIATION carry numerous discussions on current aviation news that result from an interest among wide and thorough understanding of the problems of the aeronautical industry. Other departments are: Foreign Aviation and News; Book Reviews; Aircraft in the Air; The Air Department, which is included local airport news, but a phase called by local dealers, is called by flying schools, etc. They also a section devoted to news of the U. S. Air Service. Last but not least comes Editor R. O'Brien's Slide Stops column with its interesting conclusions.

Buyers Service (Advertising)

THE ADVERTISING pages of AVIATION are a valuable and increasing source of aeronautical information. In the pages the leaders of the aeronautical industry describe their products and compare their latest developments. If you are contemplating making a flight, leaving it in purchasing a plane or any sort of aeronautical material, you will find the various pages of AVIATION invaluable.

IF YOU can use complete and authoritative information about what is going on in the aeronautical industry to advantage (as who cannot profitably and usefully) knowledge about these particular line of work) you will be interested in receiving AVIATION regularly. At twenty cents a copy in 52 issues published during a year cost more than \$10.00 and pose the risk of missing quite a lot of news. However, a subscription for a year (52 numbers) costs only \$4.95. The coupon below is to your convenience.

AVIATION

Official American Aeronautical Magazine

AVIATION PUBLISHING CORPORATION
212 West 57th Street, New York 19, N.Y.
(All orders, including with the service price, must be accompanied by payment in full.)
(Check, \$5.00; Postal, \$6.00.) (Cash if you prefer.) is authorized.

Name _____

Address _____

Phone _____

Send me _____

Enclosed is remittance for subscription

World-famous Ships of the Air—Land and Sea



AT HOME

in **San Diego**
California
Air Capital of the West

SAN DIEGO has been the scene of some most big achievements in Aviation. One of the most recent, and of which the ship has been a part, was the flight of Col. Lindbergh from San Diego to Paris in a Spirit of St. Louis, which was flown by the R. F. Mahoney Aircraft Corporation. More than 42 world records in aviation have been broken in San Diego. These records include altitude, distance, duration, and speed factors made by Army and Navy aviators in 1914. The organization was played by San Diego as the development of Aviation is also continued, in part, by the following list of facts accomplished for the first time in San Diego:

- First "Sea-Plane Flight" in the world, made by Glenn Curtiss in 1911 from San Diego Bay.
- First "Aerial Photos," taken by Major H. A. Clark in 1912.
- First "Circ. Area" (altitude) in 1, 8, 7, organized in 1912 by Maj. Victor of North Island (see Rockwell Field) San Diego.
- First "Radio Used in an Airplane," by Curtiss and Macmillan in 1913.
- First "Keep the Lamp," in an airplane by Lincoln Beachley in 1913.
- First "Night Flight," by Major T. G. Sherry, in 1913.
- First "Air Mail," San Diego in Commodore, in 1913.
- First "Aerial Building," by E. E. Smith, in 1914.
- First "Use of Plane in Coast Survey," by Rockwell Field, in 1914.

San Diego's year "round climate is certainly the most comfortable and pleasant day for "Flying" and "Sailing."

First "Transcontinental Sports Flight," in 1916.

First "Re-landing in Flight," by Smith and Barber in 1921.

First "Newest Transcontinental Flight," by McEwen and Kelly, in 1925.

First "Regular Passenger Service," San Diego to Los Angeles in 1925.

First City on the Pacific Coast to draft an "Air Ordinance," in 1925.

First "Municipal Board of Air Control" in United States in 1926.

San Diego's unparalleled natural advantages and exceptional facilities for the pursuit of aeronautics activities have been responsible for these numerous records. No other city in America, for example, has the combination of such year "round uniformly perfect climate—abundant of high winds—make supply of skilled aircraft manufacturers—ideal working conditions—and modern equipment for servicing commercial aviation and ships.

And more, "Assured" status of San Diego as leading headquarters for a Tourist airport—also in the heart of the city, and costing \$100,000.00. Designed to be one of the finest airports in the world, upon San Diego provided for business, ships, business, etc. Long-term aircraft industry will be given favorable terms on the remarkably high and so-called attractive rates. Expenses included.

Write to INDUSTRIAL DEPARTMENT, 105

Clayton of Commerce Bldg., San Diego, Calif.

FREE—An Interesting Aeronautical Book for Everyone of the Aircraft Industry.

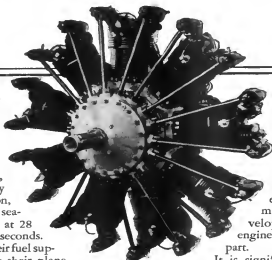


AVIATION as a weekly can present news while it is news. A news-value organization gathers information which is

A NEW ENGINE— *The Wright* CYCLONE

Makes a
New World's
Seaplane
Record

36 hours — 1 minute — 13 $\frac{1}{2}$ seconds



The Flight

WHEN, on Thursday, May 3rd, Lieutenants Gavin and Soucek, U.S.N., took off from the Delaware River at Philadelphia, in their PN-12 plane, carrying J. C. Proley, Navy mechanic, and W. F. Dayton, Wright mechanic, the world's seaplane duration record stood at 28 hours, 35 minutes, and 27 seconds.

When on May 5th, with their fuel supply exhausted, they brought their plane down, they had established a world's duration record of 36 hours, 1 minute, 13 4-5 seconds and the two Wright "Cyclone" engines had proved their fitness to carry on the Wright tradition of dependable efficiency.

The Engine

THE Wright "Cyclone" nine cylinders, 525 h. p. is a big brother to the 200 h. p. "Whirlwind" and the latest and largest member of an honored family of air-cooled engines—a family of record-makers in whose design and development the able advice of Navy engineers has played an important part.

It is significant that the Wright Aeronautical Corporation now offers for service in commercial aviation a new engine which already has a world's duration record to its credit.

*That's why
More Pilots fly them!*

WRIGHT

WRIGHT AERONAUTICAL CORPORATION, Paterson, N. J., U. S. A.
CANADIAN WRIGHT LIMITED, Sole Licensees for Canada, Montreal